

Trajectory-Tracking and Path-Following of Underactuated Autonomous Vehicles with Parametric Modeling Uncertainty

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Abstract—We address the problem of position trajectory-tracking and path-following control design for underactuated autonomous vehicles in the presence of possibly large modeling parametric uncertainty. For a general class of vehicles moving in either two or three-dimensional space, we demonstrate how adaptive switching supervisory control can be combined with a nonlinear Lyapunov-based tracking control law to solve the problem of global boundedness and convergence of the position tracking error to a neighborhood of the origin that can be made arbitrarily small. The desired trajectory does not need to be of a particular type (e.g., trimming trajectories) and can be any sufficiently smooth bounded curve parameterized by time. We also show how these results can be applied to solve the path-following problem, in which the vehicle is required to converge to and follow a path, without a specific temporal specification. We illustrate our design procedures through two vehicle control applications: a hovercraft (moving on a planar surface) and an underwater vehicle (moving in three-dimensional space). Simulations results are presented and discussed.

Index Terms—Supervisory adaptive control, path-following, trajectory-tracking, underactuated autonomous vehicles.

I. INTRODUCTION

THE past few decades have witnessed an increased research effort in the area of motion control of autonomous vehicles. A typical motion control problem is *trajectory-tracking*, which is concerned with the design of control laws that force a vehicle to reach and follow a time parameterized reference (i.e., a geometric path with an associated timing law). The degree of difficulty involved in solving this problem is highly dependent on the configuration of the vehicle. For fully actuated systems, the trajectory-tracking problem is now reasonably well understood.

For underactuated vehicles, i.e., systems with fewer actuators than degrees-of-freedom¹, trajectory-tracking is still

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¹The following definition of underactuated mechanical systems is adapted from [1], [2]. Consider the affine mechanical system described by

$$\ddot{q} = f(q, \dot{q}) + G(q)u, \quad (1)$$

an active research topic. The study of these systems is motivated by the fact that it is usually costly and often not practical to fully actuate autonomous vehicles due to weight, reliability, complexity, and efficiency considerations. Typical examples of underactuated systems include wheeled robots, hovercraft, spacecraft, aircraft, helicopters, missiles, surface vessels, and underwater vehicles. The tracking problem for underactuated vehicles is especially challenging because most of these systems are not fully feedback linearizable and exhibit nonholonomic constraints. The reader is referred to [3] for a survey of these concepts and to [4] for a framework to study the controllability and the design of motion algorithms for underactuated Lagrangian systems on Lie groups.

The classical approach for trajectory-tracking of underactuated vehicles utilizes local linearization and decoupling of the multi-variable model to steer the same number of degrees of freedom as the number of available control inputs, which can be done using standard linear (or nonlinear) control methods. Alternative approaches include the linearization of the vehicle error dynamics around trajectories that lead to a time-invariant linear system (also known as trimming trajectories) combined with gain scheduling and/or Linear Parameter Varying (LPV) design methodologies [5]–[7]. The basic limitation of these approaches is that stability is only guaranteed in a neighborhood of the selected operating points. Moreover, performance can suffer significantly when the vehicle executes maneuvers that emphasize its nonlinearity and cross-couplings. A different approach is to use output feedback linearization methods, [8]–[10]. The major challenge in this approach is that a straightforward application of this methodology, which in general involves dynamic inversion, is not always possible because certain involutivity conditions must hold [11]. In addition, even when dynamic inversion is possible, the resulting controller may not render the zero-dynamics stable.

Nonlinear Lyapunov-based designs can overcome some of the limitations mentioned above. Several examples of nonlinear trajectory-tracking controllers for marine underactuated vehicles have been reported in the literature [12]–[19]. Typically, tracking problems for autonomous vehicles are solved by designing control laws that make the vehicles track pre-

where q is a vector of independent generalized coordinates, f a vector field that captures the dynamics of the system, G the input matrix, and u the vector of generalized inputs. The system (1) is *underactuated* if the rank of G is smaller than the dimension of q , i.e., the generalized inputs are not able to instantaneously set the accelerations in all directions of the configuration space.

specified feasible “state-space” trajectories, i.e., trajectories that specify the time evolution of the position, orientation, as well as the linear and angular velocities, all consistent with the vehicles’ dynamics, [8], [13], [15]–[20], even through in practical applications one often only needs to track a desired position. This approach suffers from the drawback that usually the vehicles’ dynamics exhibit complex nonlinear terms and significant uncertainty, which makes the task of computing a feasible trajectory difficult.

It is relevant to point out that most of the results mentioned above only solve the problem in the horizontal plane. Only a few authors have tackled this control problems in three dimensional space. The reason might be that the vehicle’s dynamics become more complex and the number of degree of freedom that are not directly actuated typically increases, making the control design more involved. For example, for an underactuated underwater vehicle, the dynamics include sway and heave velocities that generate nonzero angles of sideslip and attack.

Motivated by the above considerations, we propose a solution to the trajectory-tracking problem for underactuated vehicles in both two and three-dimensional spaces. In this paper we are especially interested in situations for which there is parametric uncertainty in the model of the vehicle. Typical parameters for which this uncertainty is high, include mass and added mass for underwater vehicles which may be subject to large variations according to the payload configuration, and friction coefficients that are usually strongly dependent on the environmental conditions. The main contribution of the paper is the design of an *adaptive supervisory control* algorithm that combines logic-based switching [21] with iterative Lyapunov-based techniques such as integrator backstepping [22]. The classical approach to adaptive control relies solely on continuous tuning [22]–[24]. This approach has some inherent limitations that can be overcome by hybrid adaptive algorithms based on switching and logic [25]. The basic idea behind supervisory control [21], [26]–[30] is to design a suitable family of candidate controllers. Each controller is designed for an admissible nominal model of the process, and a supervision logic orchestrates the switching among the candidate controllers, deciding, at each instant of time, the candidate feedback controller that is more adequate. In order to guarantee stability and avoid chattering, a form of hysteresis is employed. We prove that the adaptive controller solves the problem of global boundedness and convergence of the position tracking error to a neighborhood of the origin that can be made arbitrarily small in the presence of possible large parametric uncertainty. The adaptive supervisory controller does not require persistence of excitation which sets it apart from most parameter estimation algorithms. In the control design, we take into account that the vehicle may have non-negligible dynamics and may undergo complex motions and exhibit large angles of attack and sideslip, which prevents us from using simple extensions of common control designs for wheeled robots where the total velocity vector is aligned with the vehicles main axis. Also, the desired trajectory does not need to be a trimming trajectory and can be any sufficiently smooth time-varying bounded curve, including the

degenerate case of a constant trajectory (set-point). The class of vehicles for which the design procedure is applicable is quite general and includes any vehicle modeled as a rigid-body subject to a controlled force and either one controlled torque if it is only moving on a planar surface or two or three independent control torques for a vehicle moving in three dimensional space. Furthermore, contrary to most of the approaches described above, the controller proposed does not suffer from geometric singularities due to the parameterization of the vehicle’s rotation matrix. This is possible because the attitude control problem is formulated directly in the group of rotations $SO(3)$. The literature on designing tracking control laws for underactuated vehicles directly in the configuration manifold (avoiding in this way geometric singularities) is relatively scarce. Noteworthy examples include [20], [31].

Another contribution of this paper is the application of these results to solve the *path-following* motion control problem. In path-following, the vehicle is required to converge to and follow a path that is specified without a temporal law [32]–[36]. Pioneering work in this area for wheeled mobile robots is described in [32]. In [34], Samson addressed the path-following problem for a car pulling several trailers. More recently, Altafini [36] describes a path-following controller for a n trailer vehicle that provides local asymptotic stability for a path of nonconstant curvature. Path-following controllers for aircraft and marine vehicles have been reported in [6], [9], [37]–[39]. Using the approach suggested by Hauser and Hindman [37], an output maneuvering controller was proposed in [39] for a class of strict feedback nonlinear processes and applied to path-following of fully actuated ships. The underlying assumption in path-following is that the vehicle’s forward speed tracks a desired speed profile, while the controller acts on the vehicle’s orientation to drive it to the path. Typically, in path-following, smoother convergence to the path is achieved and the control signals are less likely pushed into saturation, when compared to trajectory-tracking. In fact, in [40], [41], we highlight a fundamental difference between path-following and standard trajectory-tracking by demonstrating that performance limitations due to unstable zero-dynamics can be removed in the path-following problem. Inspired by these ideas, we solve the path-following problem by decomposing it into two subproblems: *i)* a *geometric task*, which consists of converging the vehicle to and remaining inside a tube centered around the desired path, and *ii)* a *dynamic assignment task*, which assigns a speed profile to the path.

In Section II we describe the dynamic model for the class of underactuated autonomous vehicles considered in the paper and formulate the trajectory-tracking and path-following control problems. As a preliminary material for the subsequent sections, Section III presents a nonlinear control law to solve the tracking problem and discusses the stability of the resulting closed-loop. At this point it is assumed that there is no parametric uncertainty. Sections IV and V present the main results of the paper. In Section IV, a solution to the trajectory-tracking is proposed using an estimator-based supervisory controller, and in Section V an extension is made to solve the path-following problem. In Section VI, we illustrate our

design methodologies in the context of two vehicle control applications: a hovercraft (moving on a planar surface) and an underwater vehicle (moving in three-dimensional space). The designs are validated through computer simulations. The paper concludes with a summary of the results and suggestions for further research.

A subset of the results reported here were presented in [42]–[44].

Notation: Throughout this paper, given a matrix A , A' denotes its transpose, $\lambda_{\min}(A)$, $\lambda_{\max}(A)$ are the minimum and maximum eigenvalues of A , respectively. Given two vectors $v_1 \in \mathbb{R}^{n_1}$, $v_2 \in \mathbb{R}^{n_2}$, we denote by $\text{col}(v_1, v_2)$ the vector $(v_1', v_2')' \in \mathbb{R}^{n_1+n_2}$. The Euclidean norm is denoted by $\|\cdot\|$ and the spectral norm by $\|\cdot\|_2$. A piecewise continuous function $g : [0, T) \rightarrow \mathbb{R}^n$, $T \in (0, \infty]$ is in \mathcal{L}_i , i being a positive integer, if $\int_0^T \|g(\tau)\|^i d\tau < c$ for some constant c .

II. PROBLEM STATEMENT

Consider an underactuated vehicle modeled as a rigid body subject to external forces and torques. Let $\{\mathcal{I}\}$ be an inertial coordinate frame and $\{\mathcal{B}\}$ a body-fixed coordinate frame whose origin is located at the center of mass of the vehicle. The configuration (R, p) of the vehicle is an element of the Special Euclidean group $SE(3) := SO(3) \times \mathbb{R}^3$, where $R \in SO(3) := \{R \in \mathbb{R}^{3 \times 3} : RR' = I_3, \det(R) = +1\}$ is a rotation matrix that describes the orientation of the vehicle by mapping body coordinates into inertial coordinates, and $p \in \mathbb{R}^3$ is the position of the origin of $\{\mathcal{B}\}$ in $\{\mathcal{I}\}$. Denoting by $v \in \mathbb{R}^3$ and $\omega \in \mathbb{R}^3$ the linear and angular velocities of the vehicle relative to $\{\mathcal{I}\}$ expressed in $\{\mathcal{B}\}$, respectively, the following kinematic relations apply:

$$\dot{p} = Rv \quad (2a)$$

$$\dot{R} = RS(\omega) \quad (2b)$$

where $S(\cdot)$ is a function from \mathbb{R}^3 to the space of skew-symmetric matrices $\mathcal{S} := \{M \in \mathbb{R}^{3 \times 3} : M = -M'\}$ defined by

$$S(x) := \begin{bmatrix} 0 & -x_3 & x_2 \\ x_3 & 0 & -x_1 \\ -x_2 & x_1 & 0 \end{bmatrix}, \quad \forall x := (x_1, x_2, x_3)' \in \mathbb{R}^3.$$

We consider here underactuated vehicles with dynamic equations of motion of the following form:

$$\mathbf{M}\dot{v} = -S(\omega)\mathbf{M}v + f_v(v, p, R) + g_v u_v \quad (3a)$$

$$\mathbf{J}\dot{\omega} = -S(v)\mathbf{M}v - S(\omega)\mathbf{J}\omega + f_\omega(v, \omega, p, R) + G_\omega u_\omega \quad (3b)$$

where $\mathbf{M} \in \mathbb{R}^{3 \times 3}$ and $\mathbf{J} \in \mathbb{R}^{3 \times 3}$ denote constant symmetric positive definite mass and inertia matrices; $u_v \in \mathbb{R}$ and $u_\omega \in \mathbb{R}^3$ denote the control inputs, which act upon the system through a constant nonzero vector $g_v \in \mathbb{R}^3$ and a constant nonsingular matrix² $G_\omega \in \mathbb{R}^{3 \times 3}$, respectively; the terms $-S(\omega)\mathbf{M}v$ in (3a) and $-S(v)\mathbf{M}v - S(\omega)\mathbf{J}\omega$ in (3b) are the rigid-body Coriolis terms, and the \mathcal{C}^1 functions $f_v(\cdot)$, $f_\omega(\cdot)$ represent all the remaining forces and torques acting on the body. For the special case of an underwater vehicle, \mathbf{M} and \mathbf{J} also include the so-called hydrodynamic added-mass M_A and

added-inertia J_A matrices, respectively, i.e., $\mathbf{M} = M_{RB} + M_A$, $\mathbf{J} = J_{RB} + J_A$, where M_{RB} and J_{RB} are the rigid-body mass and inertia matrices, respectively.

For an underactuated vehicle restricted to moving on a planar surface, the same equations of motion (2)–(3) apply without the first two right-hand-side terms in (3b). Also, in this case, $(R, p) \in SE(2)$, $v \in \mathbb{R}^2$, $\omega \in \mathbb{R}$, $g_v \in \mathbb{R}^2$, $G_\omega \in \mathbb{R}$, $u_\omega \in \mathbb{R}$, with all the other terms in (3) having appropriate dimensions, and the skew-symmetric matrix $S(\omega)$ is given by $S(\omega) = \begin{pmatrix} 0 & -\omega \\ \omega & 0 \end{pmatrix}$. For simplicity, in what follows, we restrict our attention to the three-dimensional case. However, all results are directly applicable to the two-dimensional case, as will be illustrated in Section VI-A for the control of a Hovercraft.

Remark 1: The vehicle dynamic model (3) does not allow f_v to depend on ω . This was done in part to simplify the analysis and also because in many vehicles this dependence is not present as is the case of the Hovercraft and the AUV described in Section VI. The methodology presented here still applies for the more general case if the dependence on ω is in the form $f_v(\cdot) = f_{v_1}(v, p, R) + f_{v_2}(v, p, R)\omega$, provided that f_{v_2} is bounded or that a suitable rank condition holds. For details see Property 1 in the Appendix and [42], [44]. \square

The problems considered in this paper can be stated as follows:

Trajectory-tracking problem: Let $p_d(t) : [0, \infty) \rightarrow \mathbb{R}^3$ be a given sufficiently smooth time-varying desired trajectory with its time-derivatives bounded. Design a controller such that all the closed-loop signals are bounded and the tracking error $\|p(t) - p_d(t)\|$ converges to a neighborhood of the origin that can be made arbitrarily small.

Path-following problem: Let $p_d(\gamma) \in \mathbb{R}^3$ be a desired path parameterized by $\gamma \in \mathbb{R}$ and $v_r(\gamma) \in \mathbb{R}$ a desired speed³ assignment. Suppose also that $p_d(\gamma)$ is sufficiently smooth with respect to γ and its derivatives (with respect to γ) are bounded. Design feedback control laws for u_v , u_ω , and $\dot{\gamma}$ such that all the closed-loop signals are bounded, the position of the vehicle converges to and remains inside a tube centered around the desired path that can be made arbitrarily thin, i.e., $\|p(t) - p_d(\gamma(t))\|$ converges to a neighborhood of the origin that can be made arbitrarily small, and the vehicle satisfies a desired speed assignment v_r along the path, i.e., the speed error $\dot{\gamma}(t) - v_r(\gamma(t))$ can be confined to an arbitrarily small ball.

III. TRAJECTORY-TRACKING CONTROLLER DESIGN

A. Controller design

This section proposes a Lyapunov-based control law to solve the trajectory-tracking problem assuming that there is no parametric uncertainty. For the sake of clarity, control-Lyapunov functions are introduced iteratively borrowing from the techniques of backstepping [22].

Step 1. Coordinate transformation: Consider the global diffeomorphic coordinate transformation

$$e := R'(p - p_d)$$

³For simplicity of presentation it will be assumed that the speed assignment $v_r(\gamma) \in \mathbb{R}$ does not depend directly on time t .

²See Remark 4 for the special case of $G_\omega \in \mathbb{R}^{3 \times 2}$.

which expresses the tracking error $p - p_d$ in the body-fixed frame. The dynamic equation of the body-fixed tracking error e is given by

$$\dot{e} = -S(\omega)e + v - R'\dot{p}_d.$$

Step 2. Convergence of e : We start by defining the control-Lyapunov function

$$V_1 := \frac{1}{2}e'e$$

and computing its time derivative to obtain

$$\dot{V}_1 = e'[v - R'\dot{p}_d]. \quad (4)$$

We can regard v as a virtual control that one would use to make \dot{V}_1 negative. This could be achieved, by setting v equal to $R'\dot{p}_d - k_e\mathbf{M}^{-1}e$, for some positive constant k_e . To accomplish this we introduce the error variable

$$z_1 := v - R'\dot{p}_d + k_e\mathbf{M}^{-1}e$$

that we would like to drive to zero, and re-write (4) as

$$\dot{V}_1 = -k_e e'\mathbf{M}^{-1}e + e'z_1. \quad (5)$$

Step 3. Backstepping for z_1 : After straightforward algebraic manipulations, the dynamic equation of the error z_1 can be written as

$$\mathbf{M}\dot{z}_1 = S(\mathbf{M}z_1)\omega + \Gamma(\cdot)\omega + g_v u_v + h(\cdot)$$

where

$$\Gamma(R, \dot{p}_d) := S(\mathbf{M}R'\dot{p}_d) - \mathbf{M}S(R'\dot{p}_d) \quad (6)$$

and $h(e, p, R, v, z_1, \ddot{p}_d) := f_v(v, p, R) - \mathbf{M}R'\ddot{p}_d + k_e z_1 - k_e^2\mathbf{M}^{-1}e$. It turns out that it will not always be possible to drive z_1 to zero. We need to explore the coupling of the translation dynamics with the rotational inputs. To this effect, we will drive z_1 to a constant design vector $\delta \in \mathbb{R}^3$. To achieve this we define $\varphi := z_1 - \delta$ as a new error variable that we will drive to zero and consider the augmented control-Lyapunov function

$$V_2 := V_1 + \frac{1}{2}\varphi'\mathbf{M}^2\varphi = \frac{1}{2}e'e + \frac{1}{2}\varphi'\mathbf{M}^2\varphi.$$

The time derivative of V_2 can be written as

$$\dot{V}_2 = -k_e e'\mathbf{M}^{-1}e + e'\delta + \varphi'(\mathbf{M}\mathbf{B}(\cdot)\zeta + \mathbf{M}h(\cdot) + e) \quad (7)$$

where

$$\begin{aligned} \mathbf{B}(R, \dot{p}_d, \delta) &:= [g_v \quad S(\mathbf{M}\delta) + \Gamma(R, \dot{p}_d)] \in \mathbb{R}^{3 \times 4} \\ \zeta &:= \text{col}(u_v, \omega) \in \mathbb{R}^4. \end{aligned} \quad (8)$$

In Appendix (cf. Property 1), we show that the matrix \mathbf{B} can always be made full-rank by choosing a suitable δ . One can now regard ζ as a virtual control (actually its first component is already a ‘‘real’’ control) that one would like to use to make \dot{V}_2 negative. This could be achieved, by setting ζ equal to

$$\alpha := \mathbf{B}'(\mathbf{B}\mathbf{B}')^{-1}(-h(\cdot) - \mathbf{M}^{-1}e - \mathbf{M}^{-1}K_\varphi\varphi),$$

where $K_\varphi \in \mathbb{R}^{3 \times 3}$ is a symmetric positive definite matrix. To accomplish this we set u_v to be equal to the first entry of α , i.e.,

$$u_v = [1 \ 0_{1 \times 3}] \alpha \quad (9)$$

and introduce the error variable

$$z_2 := \omega - [0_{3 \times 1} \ I_{3 \times 3}] \alpha$$

that one would like to set to zero. We can now re-write (7), with u_v given by (9), as

$$\dot{V}_2 = -k_e e'\mathbf{M}^{-1}e + e'\delta - \varphi'K_\varphi\varphi + \varphi'\mathbf{M}[S(\mathbf{M}\delta) + \Gamma(\cdot)]z_2.$$

Step 4. Backstepping for z_2 : Consider now a third control-Lyapunov function given by

$$V_3 := V_2 + \frac{1}{2}z_2'\mathbf{J}z_2 = \frac{1}{2}e'e + \frac{1}{2}\varphi'\mathbf{M}^2\varphi + \frac{1}{2}z_2'\mathbf{J}z_2. \quad (10)$$

Computing its time derivative one obtains

$$\begin{aligned} \dot{V}_3 &= -k_e e'\mathbf{M}^{-1}e + e'\delta - \varphi'K_\varphi\varphi + z_2'(G_\omega u_\omega - S(v)\mathbf{M}v \\ &\quad - S(\omega)\mathbf{J}\omega + f_\omega(v, \omega, p, R) - [0_{3 \times 1} \ \mathbf{J}] \dot{\alpha} \\ &\quad + [-S(\mathbf{M}\delta) + \Gamma(\cdot)']\mathbf{M}\varphi). \end{aligned}$$

For simplicity we did not expand the derivative of α . If we then choose

$$\begin{aligned} u_\omega &= G_\omega^{-1} \left(S(v)\mathbf{M}v + S(\omega)\mathbf{J}\omega - f_\omega(v, \omega, p, R) + [0_{3 \times 1} \ \mathbf{J}] \dot{\alpha} \right. \\ &\quad \left. - [-S(\mathbf{M}\delta) + \Gamma(\cdot)']\mathbf{M}\varphi - K_{z_2}z_2 \right) \end{aligned} \quad (11)$$

where $K_{z_2} \in \mathbb{R}^{3 \times 3}$ is a symmetric positive matrix, the time derivative of V_3 becomes

$$\dot{V}_3 = -k_e e'\mathbf{M}^{-1}e + e'\delta - \varphi'K_\varphi\varphi - z_2'K_{z_2}z_2.$$

Note that although \dot{V}_3 is not necessarily always negative, this will be sufficient to prove boundedness and convergence of e to a neighborhood of the origin.

B. Stability analysis

We can now prove that all signals will remain bounded, and that the tracking error converges exponential to an arbitrarily small neighborhood of the origin.

Theorem 1: Given a sufficiently smooth time-varying desired trajectory $p_d : [0, \infty) \rightarrow \mathbb{R}^3$ with its time-derivatives bounded, consider the nonlinear system Σ described by the underactuated vehicle model (2)–(3) in closed-loop with the feedback controller (9), (11).

- i) For every initial condition of Σ , the solution exists globally, all closed-loop signals are bounded, and the tracking error $\|p(t) - p_d(t)\|$ satisfies

$$\|p(t) - p_d(t)\| \leq e^{-\lambda t} c_0 + \epsilon, \quad (12)$$

where λ , c_0 , and ϵ are positive constants. From these, only c_0 depends on initial conditions.

- ii) For a given upper bound on $\|\dot{p}_d(t)\|$, by appropriate choice of the controller parameters k_e, K_φ, K_{z_2} , any desired values for ϵ and λ in (12) are possible.

Proof: To prove (i) we use Young’s inequality⁴ to conclude that for any $\gamma > 0$,

$$\begin{aligned} \dot{V}_3 &\leq -e' \left(k_e \mathbf{M}^{-1} - \frac{\gamma}{2} I \right) e - \varphi' K_\varphi \varphi \\ &\quad - z_2' K_{z_2} z_2 + \frac{1}{2\gamma} \|\delta\|^2. \end{aligned} \quad (13)$$

⁴A special case of the Young’s inequality is $ab \leq \frac{\gamma}{2} a^2 + \frac{1}{2\gamma} b^2$, where $a, b \geq 0$, and γ is any positive constant.

Suppose now that we choose γ sufficiently small so that the matrix $k_e \mathbf{M}^{-1} - \frac{\gamma}{2} \mathbf{I}$ is positive definite. In this case we conclude that there is a sufficiently small positive constant λ such that

$$\dot{V}_3 \leq -\lambda V_3 + \frac{1}{2\gamma} \|\delta\|^2, \quad (14)$$

and therefore it is straightforward to conclude from the Comparison Lemma [45] that

$$V_3(t) \leq e^{-\lambda t} V_3(0) + \frac{1}{2\lambda\gamma} \|\delta\|^2, \quad t \geq 0 \quad (15)$$

along solutions to Σ . From here we conclude that all signals remain bounded and therefore the solution exists globally. Moreover, V_3 converges to a ball of radius $\frac{1}{2\lambda\gamma} \|\delta\|^2$ and therefore $\|e\|$ converges to a ball of radius $\frac{\|\delta\|}{\sqrt{\lambda\gamma}}$, because of (10).

To prove (ii), we show next that the radius $\frac{\|\delta\|}{\sqrt{\lambda\gamma}}$ can be made as small as we want by appropriately choosing the controller parameters. To this effect, suppose we pick a desired radius ϵ and a convergence rate $\lambda > 0$, and we select δ such that \mathbf{B} is full rank. Such value for δ may depend on the upper bound of $\|\dot{p}_d(t)\|$ (see Property 1). We can then define $\gamma := \frac{\|\delta\|^2}{\epsilon^2 \lambda}$, provided that we choose k_e sufficiently large so that

$$k_e \mathbf{M}^{-1} - \frac{\gamma}{2} \mathbf{I} = k_e \mathbf{M}^{-1} - \frac{\|\delta\|^2}{2\epsilon^2 \lambda} \mathbf{I} \geq \frac{\lambda}{2} \mathbf{I} > 0.$$

If we then select $K_\varphi := \frac{\lambda}{2} \mathbf{M}^2$, $K_{z_2} := \frac{\lambda}{2} \mathbf{J}$, we conclude from (13) that (14) indeed holds for the pre-specified λ , from which (15) follows. However, now the above choices for the parameters lead to a radius $\frac{\|\delta\|}{\sqrt{\lambda\gamma}} = \epsilon$. ■

Remark 2: We did not impose any constraints on the desired trajectory (besides being sufficiently smooth and its derivative being bounded) and we also did not require that the linear velocity of the vehicle be always non-null. Consequently, $p_d(t)$ can be arbitrary, that is, the desired trajectories do not need to satisfy “dynamic” models, and in particular can be constant for all $t \geq t_0$. In that case, the controller solves the position regulation problem. □

Remark 3: In practice, the vector δ determines if the vehicle will follow the desired trajectory backwards or forwards. To observe this, define the following two angles: $\alpha = \arctan \frac{e_z}{e_x}$ and $\beta = \arctan \frac{e_y}{\sqrt{e_x^2 + e_z^2}}$, where e_x , e_y , and e_z are the three components of the body-fixed tracking error e . Notice that α and β can be seen as the elevation and azimuth angles, respectively. In steady-state (with $\dot{e} = 0$, $\dot{V}_3 = 0$), from (5), it follows that $e = \frac{1}{k_e} \mathbf{M} z_1 \approx \frac{1}{k_e} \mathbf{M} \delta$. Thus, when the first component of $\delta \in \mathbb{R}^3$ is negative and larger (in absolute value) than the other two components, the vehicle will converge to the trajectory with positive surge velocity, and will stay “behind” the desired trajectory, see examples in Section VI. □

Remark 4: When the vehicle is subject to one controlled force and only two independent control torques, i.e., $u_v \in \mathbb{R}$, but $u_\omega \in \mathbb{R}^2$ (and consequently $G_\omega \in \mathbb{R}^{3 \times 2}$), one can use, e.g., $V_3 := V_2 + \frac{1}{2} z_2' G_\omega P G_\omega' z_2$, provided that there exists a symmetric positive definite matrix P such that $P G_\omega' = G_\omega' \mathbf{J}$

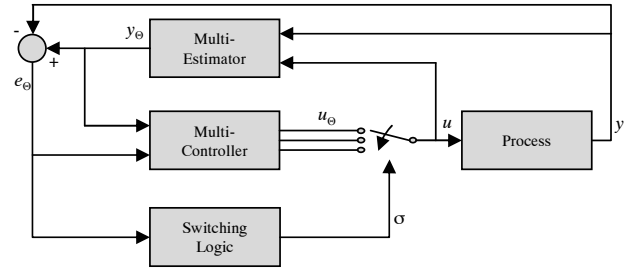


Fig. 1. Supervisory control architecture.

(which is the case for the AUV in Section VI). If we then set

$$u_\omega = (G_\omega' G_\omega)^{-1} G_\omega' \left(S(v) \mathbf{M} v + S(\omega) \mathbf{J} \omega - f_\omega(v, \omega, p, R) + [0_{3 \times 1} \ \mathbf{J}] \dot{\alpha} + [-S(\mathbf{M} \delta) + \Gamma(\cdot)'] \mathbf{M} \varphi \right) - K_{z_2} G_\omega' z_2$$

the time derivative of V_3 becomes

$$\dot{V}_3 = -k_e e' \mathbf{M}^{-1} e + e' \delta - \varphi' K_\varphi \varphi - z_2' G_\omega K_{z_2} G_\omega' z_2 + d$$

where d is a disturbance term that depends on the component of the state z_2 in the null space of G_ω' . From the above, one can prove boundedness if this component is bounded. For underwater vehicles this component typically corresponds to the roll motion which usually is stable due to the restoring forces. □

IV. ESTIMATOR-BASED SUPERVISORY CONTROL

Using the previous results, this section proposes an estimator-based supervisory control architecture to solve the trajectory-tracking problem in the presence of parametric modeling uncertainty. Let $\Theta \in \mathbb{R}^{n_\Theta}$ be a vector that contains all the unknown parameters of the dynamic equations of motion (3), where n_Θ denotes the number of unknown parameters. The following technical assumption is assumed to hold:

Assumption 1: Let \mathcal{P} be a finite set of candidate parameter values

$$\mathcal{P} := \{\Theta_1, \Theta_2, \dots, \Theta_N\}.$$

The actual parameter Θ^* belongs to \mathcal{P} . □

In practice, this assumption can be relaxed to Θ^* being sufficiently close to an element of \mathcal{P} , which can be achieved by taking a fine grid, at the price of increased computational burden.

The supervisory consists of three subsystems (see Fig. 1) [21]:

multi-estimator — a dynamical system whose inputs are the process input u and its output y , and whose outputs are \hat{y}_Θ , $\Theta \in \mathcal{P}$, where each \hat{y}_Θ is a suitably defined estimate of y which would be asymptotically correct if Θ^* was equal to Θ .

multi-controller — a dynamical system whose inputs are the output estimate \hat{y}_Θ and the estimation errors $e_\Theta := \hat{y}_\Theta - y$, $\Theta \in \mathcal{P}$, and whose outputs are the control signals u_Θ , $\Theta \in \mathcal{P}$, where each u_Θ is generated by a control law that would be adequate if Θ^* was equal to Θ .

switching logic — a dynamical system whose inputs are the estimation errors e_Θ and whose output is a switching signal σ which is used to define the control law $u = u_\sigma$.

The underlying decision-making strategy used by the switching logic basically consists of selecting for σ , the candidate controller index Θ^\dagger for which the corresponding performance signal μ_{Θ^\dagger} (which is a suitably “normed” value of e_{Θ^\dagger}) is currently the smallest. This strategy is motivated by the idea that the nominal process model with the smallest performance signal is the one that “best” approximates the actual process, and thus the candidate controller associated with that model can be expected to have a better performance of controlling the process.

In this paper we assume that the whole state of the process is available for feedback. Therefore, $y = (p, R, v, \omega)$, $\hat{y}_\Theta = (\hat{p}_\Theta, \hat{R}_\Theta, \hat{v}_\Theta, \hat{\omega}_\Theta)$, and $e_\Theta := \hat{y}_\Theta - y = (\tilde{v}_\Theta, \tilde{\omega}_\Theta)$. Since there is no uncertainty in (2), we can simply pick $\hat{p}_\Theta = p$ and $\hat{R}_\Theta = R$. We also restrict our attention to state feedback laws and therefore $u_\Theta = \text{col}(u_{v_\Theta}, u_{\omega_\Theta}) = \mathcal{K}_\Theta(\tilde{v}_\Theta, \tilde{\omega}_\Theta, p, R)$.

A. Multi-estimator

This section addresses the design of a family of estimators parameterized by $\Theta \in \mathcal{P}$ for the underactuated vehicle model (2)–(3). Motivated by Assumption 1 and in view of (3), we consider a family of estimators of the form⁵

$$\mathbf{M}_\Theta \dot{\tilde{v}}_\Theta = -S(\omega)\mathbf{M}_\Theta v + f_{v_\Theta}(v, p, R) + g_{v_\Theta} u_v - \mathbf{M}_\Theta L_v \tilde{v}_\Theta - \beta_{v_\Theta}(\cdot)\mathbf{M}_\Theta \tilde{v}_\Theta \quad (16a)$$

$$\mathbf{J}_\Theta \dot{\tilde{\omega}}_\Theta = -S(v)\mathbf{M}_\Theta v - S(\omega)\mathbf{J}_\Theta \omega + f_{\omega_\Theta}(v, \omega, p, R) + G_{\omega_\Theta} u_\omega - \mathbf{J}_\Theta L_\omega \tilde{\omega}_\Theta - \beta_{\omega_\Theta}(\cdot)\mathbf{J}_\Theta \tilde{\omega}_\Theta \quad (16b)$$

where $L_v, L_\omega \in \mathbb{R}^{3 \times 3}$ are diagonal positive definite matrices and for each $\Theta \in \mathcal{P}$ the scalar positive functions $\beta_{v_\Theta}(\tilde{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) := \beta_{1_\Theta}(\cdot) + \beta_{2_\Theta}(\cdot)$ and $\beta_{\omega_\Theta}(\tilde{v}_\Theta, \hat{\omega}_\Theta, \omega, p, R)$ satisfy⁶

$$\|f_{v_\Theta}(\hat{v}_\Theta, p, R) - f_{v_\Theta}(v, p, R)\| + \|S(\omega)\mathbf{M}_\Theta \tilde{v}_\Theta\| \leq c_1 \beta_{1_\Theta}(\hat{v}_\Theta, v, \omega, p, R) \|\tilde{v}_\Theta\| \quad (17a)$$

$$\|f_{\omega_\Theta}(\hat{v}_\Theta, \omega, p, R) - f_{\omega_\Theta}(v, \omega, p, R)\| + \|S(\hat{v}_\Theta)\mathbf{M}_\Theta - S(\mathbf{M}_\Theta v)\| \|\tilde{v}_\Theta\| + \|[0_{3 \times 1} \mathbf{J}_\Theta] \varphi_{v_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R)\| \leq c_2 \beta_{2_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) \|\tilde{v}_\Theta\| \quad (17b)$$

$$\|f_{\omega_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, p, R) - f_{\omega_\Theta}(\hat{v}_\Theta, \omega, p, R)\| + \|S(\hat{\omega}_\Theta)\mathbf{J}_\Theta - S(\mathbf{J}_\Theta \omega)\| \|\tilde{\omega}_\Theta\| + \|[0_{3 \times 1} \mathbf{J}_\Theta] \varphi_{\omega_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, \omega, p, R)\| \leq c_3 \beta_{\omega_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, \omega, p, R) \|\tilde{\omega}_\Theta\| \quad (17c)$$

for some positive constants c_i , $i = 1, \dots, 3$. The functions $\varphi_{v_\Theta}(\cdot)$ and $\varphi_{\omega_\Theta}(\cdot)$ will be defined later (cf. (28a)–(28b)). The multi-estimator has the desirable property that the estimator

⁵When \mathcal{P} has a large number of elements, an alternative approach is described in Section IV-E

⁶The existence of $\beta_{v_\Theta}(\cdot)$ and $\beta_{\omega_\Theta}(\cdot)$ follows directly from the fact that $f_{v_\Theta}(\cdot)$ and $f_{\omega_\Theta}(\cdot)$ are \mathcal{C}^1 .

error that corresponds to the actual parameter value Θ^* converges exponentially to zero and satisfies a \mathcal{L}_1 -like property.

Lemma 1: Let $\Theta^* \in \mathcal{P}$ be the actual parameter value. There exist $\kappa > 0$, $\lambda_\mu > 0$, such that for every initial condition of (2)–(3), (16), and continuous signal $u = \text{col}(u_v, u_\omega)$, there exist positive constants $\gamma_1, \gamma_2, \gamma_3$ that depend on the initial conditions such that

$$\|e_{\Theta^*}(t)\| \leq e^{-\kappa t} \gamma_1 \quad (18a)$$

$$\int_0^t e^{\lambda_\mu \tau} \beta_{v_\Theta}(\hat{v}_{\Theta^*}(\tau), \hat{\omega}_{\Theta^*}(\tau), v(\tau), \omega(\tau), p(\tau), R(\tau)) \|\tilde{v}_{\Theta^*}(\tau)\| d\tau \leq \gamma_2 \quad (18b)$$

$$\int_0^t e^{\lambda_\mu \tau} \beta_{\omega_\Theta}(\hat{v}_{\Theta^*}(\tau), \hat{\omega}_{\Theta^*}(\tau), \omega(\tau), p(\tau), R(\tau)) \|\tilde{\omega}_{\Theta^*}(\tau)\| d\tau \leq \gamma_3 \quad (18c)$$

for every time t in the maximum interval of existence of solution to the closed-loop $[0, T)$, $T \in (0, +\infty]$.

Proof: See the Appendix. ■

B. Multi-controller

We now design a family of candidate feedback laws $\mathcal{K}_\Theta(\cdot)$ such that for each $\Theta \in \mathcal{P}$, $u = \text{col}(u_v, u_\omega) = \mathcal{K}_\Theta(\cdot)$ would solve the tracking problem formulated in Section II for a process model given by (2) and (16), and “sufficiently” small estimation errors $\tilde{v}_\Theta, \tilde{\omega}_\Theta$. For a given $\Theta \in \mathcal{P}$, we design \mathcal{K}_Θ by constructing control-Lyapunov functions iteratively, following the design procedure proposed in Section III.

Step 1 and 2: Same as in Section III. However, in this case z_1 is re-defined as

$$z_{1_\Theta} := \hat{v}_\Theta - R' \dot{p}_d + k_{e_\Theta} \mathbf{M}_\Theta^{-1} e, \quad (19)$$

and therefore $\dot{V}_1 = -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' z_{1_\Theta} - e' \tilde{v}_\Theta$.

Step 3: The dynamic equation of the error z_{1_Θ} is now given by

$$\mathbf{M}_\Theta \dot{z}_{1_\Theta} = S(\mathbf{M}_\Theta z_{1_\Theta}) \omega + \Gamma_\Theta(\cdot) \omega + g_{v_\Theta} u_v + h_{1_\Theta}(\cdot) + h_{2_\Theta}(\cdot)$$

where

$$\begin{aligned} \Gamma_\Theta(R, \dot{p}_d) &:= S(\mathbf{M}_\Theta R' \dot{p}_d) - \mathbf{M}_\Theta S(R' \dot{p}_d) \\ h_{1_\Theta}(e, p, R, \hat{v}_\Theta, z_{1_\Theta}, \ddot{p}_d) &:= f_{v_\Theta}(\hat{v}_\Theta, p, R) - \mathbf{M}_\Theta R' \ddot{p}_d \\ &\quad + k_{e_\Theta} z_{1_\Theta} - k_{e_\Theta}^2 \mathbf{M}_\Theta^{-1} e \\ h_{2_\Theta}(\tilde{v}_\Theta, \hat{v}_\Theta, v, \omega, p, R) &:= -\mathbf{M}_\Theta L_v \tilde{v}_\Theta - k_{e_\Theta} \tilde{v}_\Theta \\ &\quad - \beta_{v_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) \mathbf{M}_\Theta \tilde{v}_\Theta + S(\omega) \mathbf{M}_\Theta \tilde{v}_\Theta \\ &\quad + f_{v_\Theta}(v, p, R) - f_{v_\Theta}(\hat{v}_\Theta, p, R) \end{aligned}$$

Thus, V_2 is re-defined as

$$V_2 := V_1 + \frac{1}{2} \varphi'_\Theta \mathbf{M}_\Theta^2 \varphi_\Theta = \frac{1}{2} e' e + \frac{1}{2} \varphi'_\Theta \mathbf{M}_\Theta^2 \varphi_\Theta$$

where $\varphi_\Theta := z_{1_\Theta} - \delta_\Theta$. The time derivative of V_2 can be written as

$$\begin{aligned} \dot{V}_2 &= -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta \\ &\quad + \varphi'_\Theta [\mathbf{M}_\Theta \mathbf{B}_\Theta(\cdot) \zeta_\Theta + \mathbf{M}_\Theta h_{1_\Theta}(\cdot) + h_{3_\Theta}(\cdot) + e] \end{aligned}$$

where

$$\begin{aligned} h_{3\Theta}(\tilde{v}_\Theta, \tilde{\omega}_\Theta, \hat{v}_\Theta, \hat{\omega}_\Theta, \dot{p}_d, v, \omega, p, R) &:= \mathbf{M}_\Theta h_{2\Theta}(\cdot) \\ &\quad - \mathbf{M}_\Theta [S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(\cdot)] \tilde{\omega}_\Theta \quad (20) \\ \mathbf{B}_\Theta(R, \dot{p}_d, \delta_\Theta) &:= [g_{v\Theta} \quad S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(R, \dot{p}_d)] \\ \zeta_\Theta &:= \text{col}(u_v, \hat{\omega}_\Theta). \end{aligned}$$

Following the same line of reason described in Step 3 of Section III, let

$$\alpha_\Theta := \mathbf{B}'_\Theta (\mathbf{B}_\Theta \mathbf{B}'_\Theta)^{-1} (-h_{1\Theta}(\cdot) - \mathbf{M}_\Theta^{-1} e - \mathbf{M}_\Theta^{-1} K_{\varphi_\Theta} \varphi_\Theta) \quad (21)$$

be a virtual control law for each $\Theta \in \mathcal{P}$, where δ_Θ is chosen such that $\mathbf{B}_\Theta \mathbf{B}'_\Theta$ is nonsingular. Let u_v be equal to the first entry of α_Θ , i.e.,

$$u_1 = [1 \ 0_{1 \times 3}] \alpha_\Theta \quad (22)$$

and

$$z_{2\Theta} := \hat{\omega}_\Theta - [0_{3 \times 1} \ I_{3 \times 3}] \alpha_\Theta. \quad (23)$$

Then,

$$\begin{aligned} \dot{V}_2 &= -k_{e\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta \\ &\quad + \varphi'_\Theta \mathbf{M}_\Theta [S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(\cdot)] z_{2\Theta} + \varphi'_\Theta h_{3\Theta}(\cdot). \end{aligned}$$

Step 4: The third control-Lyapunov function is now given by

$$V_3 := \frac{1}{2} e' e + \frac{1}{2} \varphi'_\Theta \mathbf{M}_\Theta^2 \varphi_\Theta + \frac{1}{2} z'_{2\Theta} \mathbf{J}_\Theta z_{2\Theta}. \quad (24)$$

Computing its time derivative one obtains

$$\begin{aligned} \dot{V}_3 &= -k_{e\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta \\ &\quad + \varphi'_\Theta h_{3\Theta}(\cdot) + z'_{2\Theta} \left[G_{\omega\Theta} u_\omega - S(v) \mathbf{M}_\Theta v - S(\omega) \mathbf{J}_\Theta \omega \right. \\ &\quad \left. + f_{\omega\Theta}(v, \omega, p, R) - \mathbf{J}_\Theta L_\omega \tilde{\omega}_\Theta - \beta_{\omega\Theta}(\cdot) \mathbf{J}_\Theta \tilde{\omega}_\Theta \right. \\ &\quad \left. - [0_{3 \times 1} \ \mathbf{J}_\Theta] \dot{\alpha}_\Theta + [-S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(\cdot)'] \mathbf{M}_\Theta \varphi_\Theta \right] \end{aligned}$$

where $\dot{\alpha}_\Theta$ can be decomposed in two terms: $\dot{\alpha}_\Theta = \hat{\dot{\alpha}}_\Theta - \check{\dot{\alpha}}_\Theta$. Here, $\check{\dot{\alpha}}_\Theta := \dot{\alpha}_\Theta - \hat{\dot{\alpha}}_\Theta$, and $\hat{\dot{\alpha}}_\Theta$ is defined to be the same as $\dot{\alpha}_\Theta$, but substituting the arguments v, ω by $\hat{v}_\Theta, \hat{\omega}_\Theta$, respectively. Selecting

$$\begin{aligned} u_\omega &= G_{\omega\Theta}^{-1} \left(S(\hat{v}_\Theta) \mathbf{M}_\Theta \hat{v}_\Theta + S(\hat{\omega}_\Theta) \mathbf{J}_\Theta \hat{\omega}_\Theta \right. \\ &\quad \left. - f_{\omega\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, p, R) + [0_{3 \times 1} \ \mathbf{J}_\Theta] \hat{\dot{\alpha}}_\Theta - [-S(\mathbf{M}_\Theta \delta_\Theta) \right. \\ &\quad \left. + \Gamma_\Theta(\cdot)'] \mathbf{M}_\Theta \varphi_\Theta - K_{z_{2\Theta}} z_{2\Theta} \right) \quad (25) \end{aligned}$$

where for each $\Theta \in \mathcal{P}$, $K_{z_{2\Theta}} \in \mathbb{R}^{3 \times 3}$ is a symmetric positive matrix, the time derivative of V_3 becomes

$$\begin{aligned} \dot{V}_3 &= -k_{e\Theta} e' \mathbf{M}_\Theta^{-1} e - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta - z'_{2\Theta} K_{z_{2\Theta}} z_{2\Theta} + e' \delta_\Theta \\ &\quad - e' \tilde{v}_\Theta + \varphi'_\Theta h_{3\Theta}(\cdot) + z'_{2\Theta} h_{4\Theta}(\cdot) \quad (26) \end{aligned}$$

where

$$\begin{aligned} h_{4\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) &:= S(\hat{v}_\Theta) \mathbf{M}_\Theta \hat{v}_\Theta - S(v) \mathbf{M}_\Theta v \\ &\quad + S(\hat{\omega}_\Theta) \mathbf{J}_\Theta \hat{\omega}_\Theta - S(\omega) \mathbf{J}_\Theta \omega + f_{\omega\Theta}(v, \omega, p, R) \\ &\quad - f_{\omega\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, p, R) - \mathbf{J}_\Theta L_\omega \tilde{\omega}_\Theta - \beta_{\omega\Theta}(\cdot) \mathbf{J}_\Theta \tilde{\omega}_\Theta \\ &\quad - [0_{3 \times 1} \ \mathbf{J}_\Theta] (\hat{\dot{\alpha}}_\Theta - \check{\dot{\alpha}}_\Theta) \quad (27) \end{aligned}$$

The last term $\dot{\alpha}_\Theta - \hat{\dot{\alpha}}_\Theta$ can be rewritten as $\dot{\alpha}_\Theta - \hat{\dot{\alpha}}_\Theta = \varphi_{v\Theta} + \varphi_{\omega\Theta}$, where

$$\begin{aligned} \varphi_{v\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) &:= \dot{\alpha}_\Theta(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) \\ &\quad - \dot{\alpha}_\Theta(\hat{v}_\Theta, \hat{\omega}_\Theta, \hat{v}_\Theta, \omega, p, R) \quad (28a) \end{aligned}$$

$$\begin{aligned} \varphi_{\omega\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, \omega, p, R) &:= \dot{\alpha}_\Theta(\hat{v}_\Theta, \hat{\omega}_\Theta, \hat{v}_\Theta, \omega, p, R) \\ &\quad - \dot{\alpha}_\Theta(\hat{v}_\Theta, \hat{\omega}_\Theta, \hat{v}_\Theta, \hat{\omega}_\Theta, p, R) \quad (28b) \end{aligned}$$

From (26), although \dot{V}_3 has indefinite terms, it will be verified that they will be dominated by the negative definite terms when the estimator errors $\tilde{v}_\Theta, \tilde{\omega}_\Theta$ are sufficiently small. This is stated in the following lemma.

Lemma 2: Let $[0, T]$, $T \in [0, \infty]$ denote the maximum interval of existence of solution to the closed-loop and suppose that there exists a time $T^\dagger \geq 0$ such that $u(t) = \mathcal{K}_{\Theta^\dagger}(\cdot)$ for all $t \in [T^\dagger, T]$ and

$$\int_{T^\dagger}^T \gamma(y(\tau), \tilde{v}_{\Theta^\dagger}(\tau), \tilde{\omega}_{\Theta^\dagger}(\tau)) d\tau < \infty \quad (29)$$

where the control law \mathcal{K}_Θ is defined in (22) and (25) and

$$\begin{aligned} \gamma(\cdot) &:= \|\mathbf{e}_{\Theta^\dagger}\|^2 + \beta_{v\Theta}(v + \tilde{v}_{\Theta^\dagger}, \omega + \tilde{\omega}_{\Theta^\dagger}, v, \omega, p, R) \|\tilde{v}_{\Theta^\dagger}\| \\ &\quad + \beta_{\omega\Theta}(v + \tilde{v}_{\Theta^\dagger}, w + \tilde{\omega}_{\Theta^\dagger}, w, p, R) \|\tilde{\omega}_{\Theta^\dagger}\|. \quad (30) \end{aligned}$$

Given a sufficiently smooth time-varying desired trajectory $p_d : [0, T] \rightarrow \mathbb{R}^3$ with its time-derivatives bounded and any initial condition of the resulting closed-loop system, the signals $e(t)$, $\hat{v}_\Theta(t)$, $\hat{\omega}_\Theta(t)$, and $u(t)$ are bounded on $[T^\dagger, T]$. Moreover, if (29) holds with $T = +\infty$, then, as $t \rightarrow \infty$, the tracking error $\|p(t) - p_d(t)\|$ converges to a neighborhood of the origin that can be made arbitrarily small by appropriate choice of the controller parameters.

Proof: See the Appendix. \blacksquare

Loosely speaking, Lemma 2 states that each candidate controller solves the trajectory-tracking problem formulated in Section II provided that the input disturbances due to the estimation errors have finite energy as defined by the integral (29). The switching-logic will guarantee that (29) holds by the Scale-Independent Hysteresis Switching Lemma [21] (cf. proof of Theorem 2).

C. Switching-logic

Motivated by (29)–(30), for each $\Theta \in \mathcal{P}$, we start by defining the performance signal μ_Θ as the state of the dynamic equation

$$\dot{\mu}_\Theta = -\lambda_\mu \mu_\Theta + \gamma(y, \tilde{v}_\Theta, \tilde{\omega}_\Theta), \quad (31)$$

with the initial values satisfying $\mu_\Theta(0) > 0$. Equation (31) implies that each performance signal μ_Θ is the sum of an exponentially decaying term that depends on initial conditions and a suitable exponentially weighted “norm” of the corresponding estimation errors. The control parameter λ_μ acts as a forgetting factor in the evaluation of the performance signals, hence establishing a compromise between adaptation alertness and switching dither.

The switching logic consider here is the *scale-independent hysteresis switching logic* proposed in [21]. Let h be a positive constant called the *hysteresis constant*. The operation of the

switching logic can be briefly explained as follows: First, we set $\sigma(0) = \arg \min_{\Theta \in \mathcal{P}} \{\mu_{\Theta}(0)\}$. Suppose that at a certain time t_i the value of σ has just switched to some $\Theta^\dagger \in \mathcal{P}$. Then, σ will be kept fixed until a time $t_{i+1} > t_i$ such that $(1+h)\min_{\Theta \in \mathcal{P}} \{\mu_{\Theta}(t_{i+1})\} \leq \mu_{\Theta^\dagger}(t_{i+1})$, at which point we set $\sigma(t_{i+1})$ to $\sigma(t_{i+1}) = \arg \min_{\Theta \in \mathcal{P}} \{\mu_{\Theta}(t_{i+1})\}$. When the indicated minimum is not unique, a particular value for σ among those that achieve the minimum can be chosen arbitrarily. Repeating this procedure, a piecewise constant signal σ is generated that is continuous from right everywhere. Setting $\mu_{\Theta}(0) > 0$ for all $\Theta \in \mathcal{P}$ avoids chattering. The switching signal is used to define the control signal as follows:

$$u = \text{col}(u_v, u_\omega) = \mathcal{K}_\sigma(\hat{v}_\Theta, \hat{\omega}_\Theta, p, R), \quad (32)$$

where the candidate control laws \mathcal{K}_Θ are defined by (22), (25).

D. Stability analysis

We are now ready to prove that all closed-loop signals will remain bounded, and that the tracking error converges to an arbitrarily small neighborhood of the origin in the presence of possible large parametric modeling uncertainty.

Theorem 2: Given a sufficiently smooth time-varying desired trajectory $p_d : [0, \infty) \rightarrow \mathbb{R}^3$ with its time-derivatives bounded, consider the hybrid system Σ_{track} described by the underactuated vehicle model (2)-(3) in closed-loop with the switched multi-controller (32), the multi-estimator (16), and the switching logic described in Section IV-C.

- i) For any initial condition of Σ_{track} with $\mu_{\Theta}(0) > 0, \forall \Theta \in \mathcal{P}$, the solution exists globally and all closed-loop signals are bounded.
- ii) Furthermore, there exists a finite time $T^\dagger \geq 0$ such that $\sigma(t) = \Theta^\dagger \in \mathcal{P}$ for all $t \geq T^\dagger$ (i.e., the switching stops in finite time) and as $t \rightarrow \infty$ the tracking error $\|p(t) - p_d(t)\|$ converges to a neighborhood of the origin that can be made arbitrarily small by appropriate choice of the control parameters.

Proof: Consider the scaled performance signals $\bar{\mu}_\Theta(t) := e^{\lambda_\mu t} \mu_\Theta(t)$, $\Theta \in \mathcal{P}$. From (31) we conclude that

$$\bar{\mu}_\Theta(t) = \bar{\mu}_\Theta(0) + \int_0^t e^{\lambda_\mu \tau} \gamma(y(\tau), \tilde{v}_\Theta(\tau), \tilde{\omega}_\Theta(\tau)) d\tau, \quad \Theta \in \mathcal{P}. \quad (33)$$

Because of the scale independence property of the switching logic, replacing μ_Θ by $\bar{\mu}_\Theta$ would have no effect on σ . From (33) we see that each $\bar{\mu}_\Theta$ is nondecreasing. This, the finiteness of \mathcal{P} , and the fact that $\bar{\mu}_\Theta(0) > 0$ for each $\Theta \in \mathcal{P}$ guarantee the existence of a positive number ϵ such that $\bar{\mu}_\Theta(t) > \epsilon, \forall t \geq 0, \forall \Theta \in \mathcal{P}$. It is not hard to conclude from the definition of the switching logic that chattering cannot occur. In fact, there must be an interval $[0, T)$ of maximal length on which the solution of the system is defined, and σ can only have a finite number of discontinuities on each proper subinterval of $[0, T)$. For details, see [21]. To prove that the switching stops in finite time, observe from (33) and (30) that $\bar{\mu}_{\Theta^\dagger}$ is bounded by virtue of Lemma 1. It follows now that the signals $\bar{\mu}_\Theta$ satisfy the hypotheses of the Scale-Independent Hysteresis Switching Lemma [21] which enables us to conclude that the switching stops in finite time. More precisely, there exists a

time $T^\dagger < T$ such that $\sigma(t) = \Theta^\dagger \in \mathcal{P}$ for all $t \in [T^\dagger, T)$. In addition, $\bar{\mu}_{\Theta^\dagger}$ is bounded on $[0, T)$. Using (33) with $\Theta = \Theta^\dagger$ and the boundedness of $\bar{\mu}_{\Theta^\dagger}$, we see that the integral

$$\int_{T^\dagger}^T \gamma(y(\tau), \tilde{v}_{\Theta^\dagger}(\tau), \tilde{\omega}_{\Theta^\dagger}(\tau)) d\tau$$

is finite (recall that λ_μ is positive). Therefore, resorting to Lemma 2, this implies that $\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, e$, and u are bounded on $[T^\dagger, T)$.

Next we will prove that \hat{v}_{Θ^\star} and $\hat{\omega}_{\Theta^\star}$ are also bounded on $[T^\dagger, T)$, where $\Theta^\star \in \mathcal{P}$ is the actual parameter value. Consider the following nonnegative function

$$V = \frac{1}{2} \hat{v}'_{\Theta^\star} \mathbf{M} \hat{v}_{\Theta^\star} + \frac{1}{2} \hat{\omega}'_{\Theta^\star} \mathbf{J} \hat{\omega}_{\Theta^\star}. \quad (34)$$

Its time derivative satisfies (c.f. Appendix)

$$\dot{V} \leq -\lambda V + [a_1(t) + a_2(t)]V + b_1(t) + b_2(t) + \kappa_1, \quad (35)$$

where $\lambda > 0$, $a_1(t), b_1(t)$ are functions in \mathcal{L}_1 defined on $[T^\dagger, T)$, and $a_2(t), b_2(t)$ are functions in \mathcal{L}_2 defined on $[T^\dagger, T)$. Consider now the ordinary differential equation

$$\begin{aligned} \dot{x}(t) = & -\lambda x(t) + [a_1(t) + a_2(t)]x(t) \\ & + b_1(t) + b_2(t) + \kappa_1, \quad t \in [T^\dagger, T) \end{aligned} \quad (36)$$

Using [46, Lemma 1] we conclude that for any initial condition, the solution $x(t)$ to (36) exists and is bounded on $[T^\dagger, T)$. Moreover, when $T = +\infty$, $x(t)$ converges to $\frac{\kappa_1}{\lambda}$ as $t \rightarrow +\infty$. Thus, applying the Comparison Lemma to (35) it can be concluded that $V(t)$ and, consequently, $\hat{v}_{\Theta^\star}, \hat{\omega}_{\Theta^\star}$ are bounded on $[T^\dagger, T)$. Since e_{Θ^\star} is bounded by virtue of Lemma 1, it follows that $(v', \omega') = (\hat{v}'_{\Theta^\star}, \hat{\omega}'_{\Theta^\star}) - e'_{\Theta^\star}$ is bounded on $[T^\dagger, T)$. Combining the boundedness of (v, ω) and $\mathcal{K}_{\Theta^\dagger}$ with the estimators (16), it can be seen that the dynamic equations for the quadratic estimation error $\|e_\Theta\|^2$, $\Theta \in \mathcal{P}$ can be expressed (after applying the Comparison Lemma) as an exponential stable linear system with bounded inputs. Therefore, this implies that e_Θ and $(\hat{v}'_\Theta, \hat{\omega}'_\Theta) = (v', \omega') + e'_\Theta$ are bounded on $[T^\dagger, T)$ for each $\Theta \in \mathcal{P}$. Since all signals are bounded in the maximal interval of existence of solutions, one concludes that the solutions exist globally, i.e., $T = +\infty$. The convergence of the tracking error $\|p(t) - p_d(t)\|$ to a neighborhood of the origin now follows from Lemma 2. ■

E. State-sharing

In the previous sections we have relied on the fact that the set \mathcal{P} was finite, so that the estimators (16). If the set \mathcal{P} is infinite or it has a large number of elements, a different approach is required. One alternative, which leads to a more efficient design, is to replace the individual estimator equations by a single system and use it to generate the estimation errors. In other words, to make the estimators in (16) “share” the same state. The performance signals μ_Θ can be obtained in a similar way. To this effect, suppose that the functions $f_{v_\Theta}(v, p, R)$, and $f_{\omega_\Theta}(v, \omega, p, R)$ are *separable* on the unknown parameter Θ , i.e., take the form $a'(v, \omega, p, R)b(\Theta)$. In that case, we can

define

$$\begin{aligned} a'_v(v, \omega, p, R, u_v) b_v(\Theta) &:= -\mathbf{M}_\Theta^{-1} S(\omega) \mathbf{M}_\Theta v \\ &\quad + \mathbf{M}_\Theta^{-1} f_{v_\Theta}(v, p, R) + \mathbf{M}_\Theta^{-1} g_{v_\Theta} u_v \\ a'_\omega(v, \omega, p, R, u_\omega) b_\omega(\Theta) &:= -\mathbf{J}_\Theta^{-1} S(v) \mathbf{M}_\Theta v - \mathbf{J}_\Theta^{-1} S(\omega) \mathbf{J}_\Theta \omega \\ &\quad + \mathbf{J}_\Theta^{-1} f_{\omega_\Theta}(v, \omega, p, R) + \mathbf{J}_\Theta^{-1} G_{\omega_\Theta} u_\omega \end{aligned}$$

and replace the estimators (16) by

$$\begin{aligned} \dot{\hat{x}}_{v_1} &= -L_v \hat{x}_{v_1} - \bar{\beta}_v(\hat{x}_v, v, \omega, p, R) \hat{x}_{v_1} + L_v v \\ &\quad + \bar{\beta}_v(\hat{x}_v, v, \omega, p, R) v \end{aligned} \quad (37a)$$

$$\begin{aligned} \dot{\hat{x}}_{v_2} &= -L_v \hat{x}_{v_2} - \bar{\beta}_v(\hat{x}_v, v, \omega, p, R) \hat{x}_{v_2} \\ &\quad + a'_v(v, \omega, p, R, u_v) \end{aligned} \quad (37b)$$

$$\begin{aligned} \dot{\hat{x}}_{\omega_1} &= -L_\omega \hat{x}_{\omega_1} - \bar{\beta}_\omega(x_v, x_\omega, \omega, p, R) \hat{x}_{\omega_1} + L_\omega \omega \\ &\quad + \bar{\beta}_\omega(x_v, x_\omega, \omega, p, R) \omega \end{aligned} \quad (37c)$$

$$\begin{aligned} \dot{\hat{x}}_{\omega_2} &= -L_\omega \hat{x}_{\omega_2} - \bar{\beta}_\omega(x_v, x_\omega, \omega, p, R) \hat{x}_{\omega_2} \\ &\quad + a'_\omega(v, \omega, p, R, u_\omega) \end{aligned} \quad (37d)$$

with outputs

$$\hat{v}_\Theta = \hat{x}_{v_1} + \hat{x}_{v_2} b_v(\Theta), \quad \hat{\omega}_\Theta = \hat{x}_{\omega_1} + \hat{x}_{\omega_2} b_\omega(\Theta)$$

where

$$\begin{aligned} \bar{\beta}_v(\hat{x}_v, v, \omega, p, R) &= \sup_{\Theta} \beta_{v_\Theta}(\hat{x}_{v_1} + \hat{x}_{v_2} b_v(\Theta), \hat{x}_{\omega_1} \\ &\quad + \hat{x}_{\omega_2} b_\omega(\Theta), v, \omega, p, R) \\ \bar{\beta}_\omega(x_v, x_\omega, \omega, p, R) &= \sup_{\Theta} \beta_{\omega_\Theta}(\hat{x}_{v_1} + \hat{x}_{v_2} b_v(\Theta), \hat{x}_{\omega_1} \\ &\quad + \hat{x}_{\omega_2} b_\omega(\Theta), \omega, p, R). \end{aligned}$$

Note that \hat{v}_Θ and $\hat{\omega}_\Theta$ satisfy (16) but with possible larger β_{v_Θ} and β_{ω_Θ} that still satisfy (17). However, the dimension of (37) is now independent of the number of elements in \mathcal{P} .

V. PATH-FOLLOWING CONTROLLER DESIGN

In this section, inspired by [39], the results described in Section IV are utilized to solve the path-following problem. Let $p_d(\gamma) \in \mathbb{R}^3$ be a desired geometric path parameterized by a variable $\gamma \in \mathbb{R}$ and $v_r(\gamma) \in \mathbb{R}$ a desired speed assignment. Contrary to trajectory-tracking, in path-following we have the freedom to select a timing law for $\gamma(t)$. In particular, we can regard $\gamma(t)$ as an additional control input. In this paper, we actually regard $\dot{\gamma}(t)$ as the additional input, because this will necessarily produce a differentiable $\gamma(t)$. Let us define the position body-fixed path-following error $e := R'[p(t) - p_d(\gamma(t))]$ and the speed error $z_\gamma := \dot{\gamma} - v_r$. Following the same steps described in Section III and IV-B, and defining z_{1_Θ} (see (19)) as $z_{1_\Theta} := \hat{v}_\Theta - R' p_d^\gamma v_r + k_{e_\Theta} \mathbf{M}_\Theta^{-1} e$, where $p_d^\gamma := \frac{\partial p_d}{\partial \gamma}$, we obtain

$$\dot{V}_1 = -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' z_{1_\Theta} - e' \tilde{v}_\Theta - e' R' p_d^\gamma z_\gamma.$$

Notice also that

$$\begin{aligned} \mathbf{M}_\Theta \dot{z}_{1_\Theta} &= S(\mathbf{M}_\Theta z_{1_\Theta}) \omega + \Gamma_\Theta(\cdot) \omega + g_{v_\Theta} u_v \\ &\quad + h_{1_\Theta}(\cdot) + h_{2_\Theta}(\cdot) + h_{5_\Theta}(\cdot) z_\gamma \end{aligned}$$

where

$$\Gamma_\Theta(R, p_d^\gamma, v_r) := S(\mathbf{M}_\Theta R' p_d^\gamma v_r) - \mathbf{M}_\Theta S(R' p_d^\gamma v_r)$$

$$\begin{aligned} h_{1_\Theta}(e, p, R, \hat{v}_\Theta, z_{1_\Theta}, p_d^\gamma, p_d^{\gamma^2}, v_r, v_r^\gamma) &:= f_{v_\Theta}(\hat{v}_\Theta, p, R) \\ &\quad - \mathbf{M}_\Theta R'(p_d^{\gamma^2} v_r^2 + p_d^\gamma v_r^\gamma v_r) + k_{e_\Theta} z_{1_\Theta} - k_{e_\Theta}^2 \mathbf{M}_\Theta^{-1} e \\ h_{2_\Theta}(\tilde{v}_\Theta, \hat{v}_\Theta, v, \omega, p, R) &:= -\mathbf{M}_\Theta L_v \tilde{v}_\Theta \\ &\quad - \beta_{v_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) \mathbf{M}_\Theta \tilde{v}_\Theta - k_{e_\Theta} \tilde{v}_\Theta + S(\omega) \mathbf{M}_\Theta \tilde{v}_\Theta \\ &\quad + f_{v_\Theta}(v, p, R) - f_{v_\Theta}(\hat{v}_\Theta, p, R) \\ h_{5_\Theta}(R, p_d^\gamma, p_d^{\gamma^2}, v_r, v_r^\gamma) &:= -\mathbf{M}_\Theta R'(p_d^{\gamma^2} v_r + p_d^\gamma v_r^\gamma) - k_{e_\Theta} R' p_d^\gamma \\ \text{and } p_d^{\gamma^2} &:= \frac{\partial^2 p_d}{\partial \gamma^2}, v_r^\gamma := \frac{\partial v_r}{\partial \gamma}. \text{ Therefore, using (21)–(23), we} \\ \text{obtain} \end{aligned}$$

$$\begin{aligned} \dot{V}_2 &= -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta \\ &\quad + \varphi'_\Theta \mathbf{M}_\Theta [S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta] z_{2_\Theta} \\ &\quad + \varphi'_\Theta h_{3_\Theta}(\cdot) + [-e' R' p_d^\gamma + \varphi'_\Theta \mathbf{M}_\Theta h_{5_\Theta}(\cdot)] z_\gamma \end{aligned}$$

where

$$\begin{aligned} h_{3_\Theta}(\tilde{v}_\Theta, \tilde{\omega}_\Theta, \hat{v}_\Theta, p_d^\gamma, v_r, v, \omega, p, R) &:= \mathbf{M}_\Theta h_{2_\Theta}(\cdot) \\ &\quad - \mathbf{M}_\Theta [S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(\cdot)] \tilde{\omega}_\Theta \\ B_\Theta(R, p_d^\gamma, v_r, \delta_\Theta) &:= [g_{v_\Theta}, S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(R, p_d^\gamma, v_r)] \\ \zeta_\Theta &:= \text{col}(u_v, \hat{\omega}_\Theta) \end{aligned}$$

Notice that $\dot{\alpha}_\Theta$ can be decomposed as $\dot{\alpha}_\Theta = h_{6_\Theta}(\cdot) + h_{7_\Theta}(\cdot) z_\gamma$. Thus, if we then choose

$$\begin{aligned} u_\omega &= G_{\omega_\Theta}^{-1} \left(S(\hat{v}_\Theta) \mathbf{M}_\Theta \hat{v}_\Theta + S(\hat{\omega}_\Theta) \mathbf{J}_\Theta \hat{\omega}_\Theta - f_{\omega_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, p, R) \right. \\ &\quad + [0_{3 \times 1} \mathbf{J}_\Theta] \hat{h}_{6_\Theta}(\cdot) - [-S(\mathbf{M}_\Theta \delta_\Theta) + \Gamma_\Theta(\cdot)'] \mathbf{M}_\Theta \varphi_\Theta \\ &\quad \left. - K_{z_{2_\Theta}} z_{2_\Theta} \right), \end{aligned} \quad (38)$$

the time derivative of V_3 becomes

$$\begin{aligned} \dot{V}_3 &= -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta \\ &\quad + \varphi'_\Theta h_{3_\Theta}(\cdot) - z'_{2_\Theta} K_{z_{2_\Theta}} z_{2_\Theta} + z'_{2_\Theta} h_{4_\Theta}(\cdot) \\ &\quad - (e' R' p_d^\gamma - \varphi'_\Theta \mathbf{M}_\Theta h_{5_\Theta}(\cdot) + z'_{2_\Theta} [0_{3 \times 1} \mathbf{J}_\Theta] \hat{h}_{7_\Theta}(\cdot)) z_\gamma \end{aligned}$$

where $\tilde{h}_{i_\Theta} := \hat{h}_{i_\Theta} - h_{i_\Theta}$, $i = 6, 7$, \hat{h}_{i_Θ} is defined to be the same as h_{i_Θ} , but substituting v, ω by $\hat{v}_\Theta, \hat{\omega}_\Theta$, and

$$\begin{aligned} h_{4_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, v, \omega, p, R) &:= S(\hat{v}_\Theta) \mathbf{M}_\Theta \hat{v}_\Theta - S(v) \mathbf{M}_\Theta v \\ &\quad + S(\hat{\omega}_\Theta) \mathbf{J}_\Theta \hat{\omega}_\Theta - S(\omega) \mathbf{J}_\Theta \omega + f_{\omega_\Theta}(v, \omega, p, R) \\ &\quad - f_{\omega_\Theta}(\hat{v}_\Theta, \hat{\omega}_\Theta, p, R) - \mathbf{J}_\Theta L_\omega \tilde{\omega}_\Theta + \beta_{\omega_\Theta}(\cdot) \mathbf{J}_\Theta \tilde{\omega}_\Theta \\ &\quad + [0_{3 \times 1} \mathbf{J}_\Theta] (\tilde{h}_{6_\Theta}(\cdot) + \tilde{h}_{7_\Theta}(\cdot) z_\gamma) \end{aligned}$$

Introduce now a forth control Lyapunov function given by

$$V_4 := V_3 + \frac{1}{2} z_\gamma^2 = \frac{1}{2} e' e + \frac{1}{2} \varphi'_\Theta \mathbf{M}_\Theta^2 \varphi_\Theta + \frac{1}{2} z'_{2_\Theta} \mathbf{J}_\Theta z_{2_\Theta} + \frac{1}{2} z_\gamma^2.$$

Computing its time derivative, we get

$$\begin{aligned} \dot{V}_4 &= -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta + \varphi'_\Theta h_{3_\Theta}(\cdot) \\ &\quad - z'_{2_\Theta} K_{z_{2_\Theta}} z_{2_\Theta} + z'_{2_\Theta} h_{4_\Theta}(\cdot) + z_\gamma (-e' R' p_d^\gamma \\ &\quad + \varphi'_\Theta \mathbf{M}_\Theta h_{5_\Theta}(\cdot) - z'_{2_\Theta} [0_{3 \times 1} \mathbf{J}_\Theta] \hat{h}_{7_\Theta}(\cdot) + \dot{\gamma} - v_r^\gamma \dot{\gamma}). \end{aligned}$$

Selecting the following update law for $\dot{\gamma}$:

$$\begin{aligned} \dot{\gamma} &= e' R' p_d^\gamma - \varphi'_\Theta \mathbf{M}_\Theta h_{5_\Theta}(\cdot) + z'_{2_\Theta} [0_{3 \times 1} \mathbf{J}_\Theta] \hat{h}_{7_\Theta}(\cdot) \\ &\quad + v_r^\gamma \dot{\gamma} - k_\gamma z_\gamma \end{aligned} \quad (39)$$

where k_γ is a positive constant, we obtain

$$\dot{V}_4 = -k_{e_\Theta} e' \mathbf{M}_\Theta^{-1} e + e' \delta_\Theta - e' \tilde{v}_\Theta - \varphi'_\Theta K_{\varphi_\Theta} \varphi_\Theta$$

$$+\varphi'_\Theta h_{3\Theta}(\cdot) - z'_{2\Theta} K_{z_{2\Theta}} z_{2\Theta} + z'_{2\Theta} h_{4\Theta}(\cdot) - k_\gamma z_\gamma^2.$$

An extension of Theorem 2 to the path-following then follows.

Theorem 3: Given a sufficiently smooth (with respect to γ) desired path $p_d(\gamma) : \mathbb{R} \rightarrow \mathbb{R}^3$ with its derivatives (with respect to γ) bounded, and a desired assignment speed $v_r(\gamma) : \mathbb{R} \rightarrow \mathbb{R}$, consider the hybrid system Σ_{path} described by the underactuated vehicle model (2)–(3) in closed-loop with the switched multi-controller (22), (38), (39), the multi-estimator (16), and the switching logic described in Section IV-C.

- i) For any initial condition of Σ_{path} with $\mu_\Theta(0) > 0$, $\forall \Theta \in \mathcal{P}$, the solution exists globally and all closed-loop signals are bounded.
- ii) There exist a finite time $T^\dagger \geq 0$ such that $\sigma(t) = \Theta^\dagger \in \mathcal{P}$ for all $t \geq T^\dagger$ (i.e., the switching stops in finite time). Moreover, as $t \rightarrow \infty$, the position error $\|p(t) - p_d(\gamma(t))\|$ and the speed error $\|\dot{\gamma}(t) - v_r(\gamma(t))\|$ converge to neighborhoods of the origin that can be made arbitrarily small by appropriate choice of the control parameters.

Proof: The proof is not given since it is a simple application of the arguments used in the previous theorems. ■

VI. APPLICATION TO SPECIFIC VEHICLES

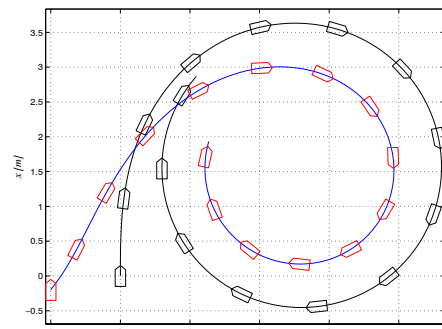
This section illustrates the application of the previous results to two vehicles: a hovercraft (moving on a planar surface) and an underwater vehicle (moving in three-dimensional space).

A. Trajectory-tracking of an underactuated Hovercraft

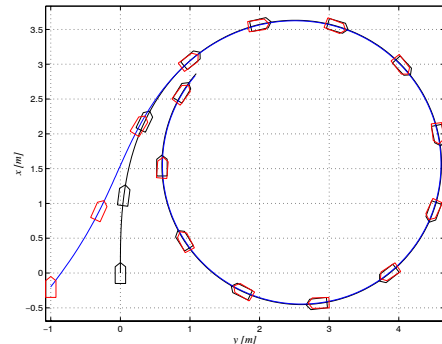
Consider the Caltech MVWT vehicle described in [43], [47] consisting of a platform mounted on three low-friction, omnidirectional casters, with two attached high-performance ducted fans. Let $p = (x, y)' \in \mathbb{R}^2$ be the Cartesian coordinates of the vehicle's center of mass and $\theta \in \mathbb{S}^1$ its orientation. Assuming that the friction and moment forces can be modeled by viscous friction, the equations of motion are

$$\begin{aligned} m\ddot{x} &= -d_v \dot{x} + (F_s + F_p) \cos \theta \\ m\ddot{y} &= -d_v \dot{y} + (F_s + F_p) \sin \theta \\ J\ddot{\theta} &= -d_\omega \dot{\theta} + l(F_s - F_p) \end{aligned}$$

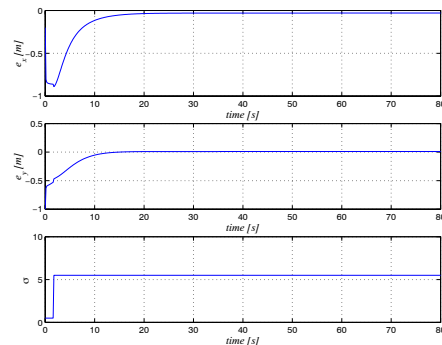
where $m = 5.5 \text{ kg}$ is the mass of the vehicle and $J = 0.047 \text{ Kg m}^2$ is the rotational inertia. The starboard and portboard fan forces are denoted F_s and F_p , respectively, and $l = 0.123 \text{ m}$ denotes the moment arm of the forces. The geometric and mass centers of the vehicle are assumed to coincide. The coefficient of viscous friction d_v is 5.5 Kg/s and the coefficient of rotational friction d_ω is 0.41 Kg m/s . Expressing the equations of motion in the body fixed frame, yields (2)–(3) with $(R, p) \in SE(2)$, $v \in \mathbb{R}^2$, $\omega \in \mathbb{R}$, $R(\theta) = \begin{pmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{pmatrix}$, $S(\omega) = \begin{pmatrix} 0 & -\omega \\ \omega & 0 \end{pmatrix}$, $\mathbf{M} = \text{diag}\{m, m\}$, $\mathbf{J} = J$, $f_v = -\text{diag}\{d_v, d_v\}v$, $f_\omega = -d_\omega \omega$, $g_v = (1, 0)'$, $G_\omega = 1$, $u_v = F_s + F_p$, and $u_\omega = l(F_s - F_p)$. In this case the matrix \mathbf{B} introduced in (8) is given by $\mathbf{B} = \begin{pmatrix} 1 & m\delta_2 \\ 0 & -m\delta_1 \end{pmatrix}$ with $\delta = (\delta_1, \delta_2)'$. The reader is referred to [43] for a detailed coverage of the trajectory-tracking controller with experimental results.



(a)



(b)



(c)

Fig. 2. First experiment. Trajectory of the hovercraft in the xy -plane and reference trajectory performed by a unicycle vehicle using the trajectory-tracking controller presented in Section III (diagram (a)) and the estimator-based supervisory controller for trajectory-tracking (diagram (b)). Time evolution of the tracking error in x -direction, in y -direction, and the switching signal σ for the estimator-based supervisory controller (diagram (c)).

We now describe two simulation results that illustrate the performance of the proposed tracking controller with and without supervisory control. The objective of the first experiment is to force the hovercraft to track the “virtual” kinematic unicycle vehicle

$$\dot{x}_d = V_d \cos \theta_d, \quad \dot{y}_d = V_d \sin \theta_d, \quad \dot{\theta}_d = \omega_d,$$

which starts at $x_d(0) = y_d(0) = \theta_d(0) = 0$ and moves with velocities $V_d(t) = 0.2 \text{ m/s}$ and $\omega_d(t) = 0.1 \text{ rad/s}$. The initial conditions for the hovercraft are $(x_0, y_0) = (-0.2 \text{ m}, -1 \text{ m})$, $R_0 = I$, $v_0 = \omega_0 = 0$. For simplicity, only the coefficient of viscous friction is unknown, but assumed to belong to the set $\mathcal{P} = \{0.5, 1.0, \dots, 9.5, 10.0\}$. The control parameters were

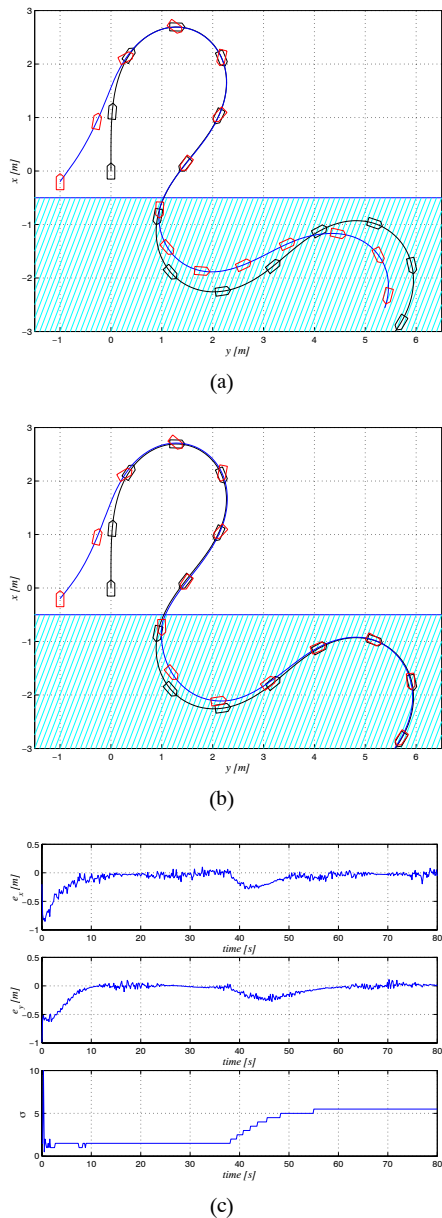


Fig. 3. Second experiment. Trajectory of the hovercraft in the xy -plane and reference trajectory performed by a unicycle vehicle using the trajectory-tracking controller presented in Section III (diagram (a)) and the estimator-based supervisory controller for trajectory-tracking (diagram (b)). Time evolution of the tracking error in x -direction, in y -direction, and the switching signal σ for the estimator-based supervisory controller (diagram (c)).

selected as follows: $k_{e_\theta} = 1.5$, $K_{\varphi_\theta} = 8I$, $K_{z_{2\theta}} = 8$, and $\delta_\theta = (-0.01, 0)'$ for all $\theta \in \mathcal{P}$. The hysteresis constant for the switching logic was set to $h = 0.1$, the forgetting factor to $\lambda_\mu = 1.0$, and the multi-estimator gains to $L_v = 0.1I$ and $L_\omega = 0.1$. The functions $\beta_v(\cdot)$ and $\beta_\omega(\cdot)$ introduced in (16) are given by $\beta_v(\cdot) = \beta_\omega(\cdot) = \omega^2$.

To illustrate the benefits derived from the supervisory control scheme proposed in Section IV, we show in Fig. 2(a) the closed-loop trajectory for the (non-adaptive) trajectory-tracking controller presented in Section III when the value of the coefficient of viscous friction assumed by the control system was set to 10% of the real value. It can be seen that

although the closed-loop is still stable, the parameter error affects considerably the closed-loop performance. In contrast, Fig. 2(b) shows the closed-loop trajectory for the supervisory controller where, as expected, the hovercraft converges to a small neighborhood of the “virtual” unicycle vehicle, in spite of the uncertainty in d_v . Fig. 2(c) shows the time evolution of some relevant variables. In steady-state the vehicle is not aligned with the direction of the tangent velocity of p_d . Contrary to what happens for wheeled mobile robots (with inherent lateral drag coefficient $d_v = +\infty$) in the hovercraft case we cannot force the orientation θ to converge to the direction of the tangent velocity p_d .

To further illustrate the usefulness of the adaptive scheme and test its robustness with respect to sensor noise, a second experiment is described. In this case, all the initial conditions and control parameters are as in the first experiment, but now the “virtual” unicycle vehicle moves with linear velocity $V_d = 0.2 \text{ m/s}$ and angular velocity ω_d such that

$$\dot{\omega}_d = -0.1(\omega_d + 0.3 \sin(t/8)), \quad \omega_d(0) = 0.$$

Zero mean uniform random noise was introduced in every sensed signal: the measured velocities v , and ω ; the orientation angle θ ; and the x and y positions. The amplitude was set to $(0.05, 0.05)$, 0.05 , 0.1 , 0.1 , and 0.1 , respectively. We also consider the situation where the Hovercraft moves between two surfaces characterized by distinct friction coefficients (e.g., from water to land). To simulate this effect, we set the value of the coefficient of viscous friction to $d_v = 1.5 \text{ Kg/s}$ while the Hovercraft is in the region $\{(x, y) \in \mathbb{R}^2 : x \geq -0.5\}$ and $d_v = 5.5 \text{ Kg/s}$, otherwise. We can see in Fig. 3(b) that the hovercraft still converges to a very small neighborhood of the target unicycle vehicle and its performance is not significantly affected by the switching in d_v .

B. Trajectory-tracking and path-following of an underwater vehicle in 3-D space

Consider an ellipsoidal shaped underactuated autonomous underwater vehicle (AUV) not necessarily neutrally buoyant. Let $\{\mathcal{B}\}$ be a body-fixed coordinate frame whose origin is located at the center of mass of the vehicle and suppose that we have available a pure body-fixed control force τ_u in the x_B direction, and two independent control torques τ_q and τ_r about the y_B and z_B axes of the vehicle, respectively. The kinematics and dynamics equations of motion of the vehicle can be written as (2)–(3), where $\mathbf{M} = \text{diag}\{m_{11}, m_{22}, m_{33}\}$, $\mathbf{J} = \text{diag}\{J_{11}, J_{22}, J_{33}\}$, $u_v = \tau_u$, $u_\omega = (\tau_q, \tau_r)'$, $D_v(v) = \text{diag}\{X_{v_1} + X_{|v_1|v_1}|v_1|, Y_{v_2} + Y_{|v_2|v_2}|v_2|, Z_{v_3} + Z_{|v_3|v_3}|v_3|\}$, $D_\omega(\omega) = \text{diag}\{K_{\omega_1} + K_{|\omega_1|\omega_1}|\omega_1|, M_{\omega_2} + M_{|\omega_2|\omega_2}|\omega_2|, N_{\omega_3} + N_{|\omega_3|\omega_3}|\omega_3|\}$, $g_v = \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix}$, $G_\omega = \begin{pmatrix} 0 & 0 \\ 0 & 1 \\ 0 & 0 \end{pmatrix}$, $\bar{g}_1(R) = R' \begin{pmatrix} 0 \\ 0 \\ W-B \end{pmatrix}$, $\bar{g}_2(R) = S(r_B)R' \begin{pmatrix} 0 \\ 0 \\ B \end{pmatrix}$, $f_v = -D_v(v)v - \bar{g}_1(R)$, $f_\omega = -D_\omega(\omega)\omega - \bar{g}_2(R)$. The gravitational and buoyant forces are given by $W = mg$ and $B = \rho g \nabla$, respectively, where m is the mass, ρ is the mass density of the water and ∇ is the volume of the displaced water. The numerical values used for the physical parameters match those of the *Sirene* AUV, described in [48], [49]. The matrix B defined in (8)

takes the form $B = [b_1, b_2, b_3, b_4]$, where $b_1 := [1, 0, 0]'$, $b_2 := [0, m_{33}\delta_3 + a_3(m_{33} - m_{22}), -m_{22}\delta_2 - a_2(m_{22} - m_{33})]'$, $b_3 := [-m_{33}\delta_3 - a_3(m_{33} - m_{11}), 0, m_{11}\delta_1 + a_1(m_{11} - m_{33})]'$, $b_4 := [m_{22}\delta_2 + a_2(m_{22} - m_{11}), -m_{11}\delta_1 + a_1(m_{11} - m_{22}), 0]'$, $a = (a_1, a_2, a_3)' := R'\dot{p}_d$.

Two simulation results are included to illustrate the dynamic behavior of the AUV in closed-loop with the trajectory-tracking controller presented in Section III, and the path-following controller in Section V. Fig. 4(a) displays the vehicle trajectory using the trajectory-tracking controller in the 3D-space for the following desired trajectory

$$p_d(t) = [V_1 \cos(\frac{2\pi}{T}t + \phi_d), V_1 \sin(\frac{2\pi}{T}t + \phi_d), V_2 t]$$

with $V_1 = 20$, $V_2 = 0.05$, $T = 400$, and $\phi_d = -\frac{\pi}{2}$. The initial conditions of the AUV are $p_0 = (x_0, y_0, z_0) = (10m, -10m, 0)$, $R = I$, and $v_0 = \omega_0 = 0$. The control parameters were selected as follows: $k_e = 0.1$, $K_\varphi = I$, $K_{z_2} = I$, and $\delta = (-2 \max(|m_{11} - m_{22}|, |m_{11} - m_{33}|), 0, 0)'$. Figures 4(c) and 4(e) show the time evolution of the tracking error e and the Euler angles (computed from R), respectively. The damped oscillatory behavior of pitch and roll are due to the gravitational and buoyancy forces. Notice that the initial position of the desired position was deliberately chosen to be almost behind the initial position of the vehicle. As we can see, the vehicle turns back in its attempt to be at the given reference position at the prescribed time, requiring significant control effort and consequently inducing a strong oscillatory behavior. The path-following controller in Section V was used to generate the trajectories in Fig. 4(b), 4(d), and 4(f) where the desired path, the initial conditions and the control gains are the same as in the experiment for the tracking controller. The guidance gain and the speed assignment were set to $k_\gamma = 2$ and $v_r = 1m/s$, respectively. The initial condition for γ was chosen to be the one that minimizes the distance between the initial position of the AUV and the desired path. The convergence of the vehicle to the path is now much smoother. From these two experiments one can see that when the primary objective is to steer the vehicle to converge to and move along a geometric path, the path-following controller offers significant performance improvement. For simplicity, in these experiments we did not include model uncertainty.

VII. CONCLUSIONS

We proposed a solution to the trajectory-tracking and path-following problem for underactuated autonomous vehicles in the presence of possibly large modeling parametric uncertainty. For a general class of vehicles moving in either two or three-dimensional space, we demonstrated how adaptive switching supervisory control can be combined with a nonlinear Lyapunov-based tracking control law to design a hybrid controller that yields global boundedness and convergence of the position tracking error to a small neighborhood, and robustness to parametric modeling uncertainty. We illustrated our results in the context of two vehicle control applications: a hovercraft (moving on a planar surface) and an underwater vehicle (moving in three-dimensional space). Simulations show that the control objectives were accomplished.

An alternative approach to the Lyapunov-based control scheme proposed in Section III consists in choosing an adequate point linked to the vehicle and then utilize output feedback linearization to design a simple controller that drives that point to the reference trajectory. See for example [10] for the case of unicycle-like mobile robots. For general underactuated vehicles, the stability of the zero-dynamics would have to be established independently. This is an issue for future research.

A problem that warrants further research is the control of underactuated vehicles with noise and in the presence of disturbances. Typical disturbances for marine vehicles include the ones induced by wave, wind, and ocean current.

APPENDIX

Property 1: Let $g_v = (g_{v_1}, g_{v_2}, g_{v_3})'$ be non-zero and \dot{p}_d uniformly bounded by $\|\dot{p}_d\| \leq \gamma_{\dot{p}_d}$. Then, there exists a vector δ that makes $B(R, \dot{p}_d, \delta)$ defined in (8) full-rank.

Proof: Pick $\delta = M^{-1}\mu\varepsilon$, where $\mu = (\mu_1, \mu_2, \mu_3)' := (1, 0, 0)'$ and ε is a positive constant to be selected shortly. Defining $\Delta := \frac{1}{\varepsilon}\Gamma$, we conclude that

$$\begin{aligned} \text{rank } B &= \text{rank} [g_v \ \varepsilon S(\mu) + \Gamma] \begin{bmatrix} I & 0 \\ 0 & \frac{1}{\varepsilon}I \end{bmatrix} \\ &= \text{rank} [g_v \ S(\mu) + \Delta] \\ &= \text{rank} \begin{bmatrix} g_{v_1} & \Delta_{11} & -\mu_3 + \Delta_{12} & \mu_2 + \Delta_{13} \\ g_{v_2} & \mu_3 + \Delta_{21} & \Delta_{22} & -\mu_1 + \Delta_{23} \\ g_{v_3} & -\mu_2 + \Delta_{31} & \mu_1 + \Delta_{32} & \Delta_{33} \end{bmatrix} \end{aligned} \quad (40)$$

where we used the fact that the rank of a matrix does not change when it is multiplied by a non-singular matrix. From the definition of Γ (see (6)) we conclude that each element of Δ can be bounded by $|\Delta_{ij}| \leq \frac{2\lambda_{\max}(M)\gamma_{\dot{p}_d}}{\varepsilon}$. Much tighter bounds can be obtained for specific class of vehicles. For example when $M = mI$ then $\Gamma = \Delta = 0$. Assume now that $g_{v_1} \neq 0$. From (40), we conclude that B has full-rank if one can find at least one 3×3 minor determinant of $[g_v \ S(\mu) + \Delta]$ non-zero. Consider the one formed by the first, third and fourth columns, i.e.,

$$\begin{aligned} \begin{vmatrix} g_{v_1} & \Delta_{12} & \Delta_{13} \\ g_{v_2} & \Delta_{22} & -\mu_1 + \Delta_{23} \\ g_{v_3} & \mu_1 + \Delta_{32} & \Delta_{33} \end{vmatrix} &= \mu_1 [\mu_1 g_{v_1} + g_{v_1} \Delta_{32} - g_{v_1} \Delta_{23} \\ &\quad + g_{v_2} \Delta_{13} - g_{v_3} \Delta_{12}] + h_\Delta \end{aligned} \quad (41)$$

where h_Δ collects all the remaining terms and does not depend on μ_1 . It follows now from the fact that each Δ_{ij} can be arbitrarily small by choosing ε (and consequently δ) sufficient large, that (41) can be made nonzero and therefore B full-rank. If $g_{v_1} = 0$, the same conclusion about the full-rank of B can be made by following the same reasoning but with $\mu = (0, 1, 0)'$ if $g_{v_2} \neq 0$, or $\mu = (0, 0, 1)'$ for the case $g_{v_1} = g_{v_2} = 0$. ■

Lemma 1

Proof: Throughout this proof, to avoid cumbersome notation, we will use \hat{v} , \tilde{v} , $\hat{\omega}$, $\tilde{\omega}$ to denote \hat{v}_{Θ^*} , \tilde{v}_{Θ^*} , $\hat{\omega}_{\Theta^*}$, and $\tilde{\omega}_{\Theta^*}$, respectively. Note also that $\Theta = \Theta^*$ corresponds to the nominal model and therefore M_{Θ^*} is equal to M . The same applies for the other model parameters. Consider the following exponentially weighted Lyapunov-like functions

$$V_v := \frac{1}{2} e^{2\lambda_\mu t} \|\tilde{v}\|^2, \quad V_\omega := \frac{1}{2} e^{2\lambda_\mu t} \|\tilde{\omega}\|^2$$

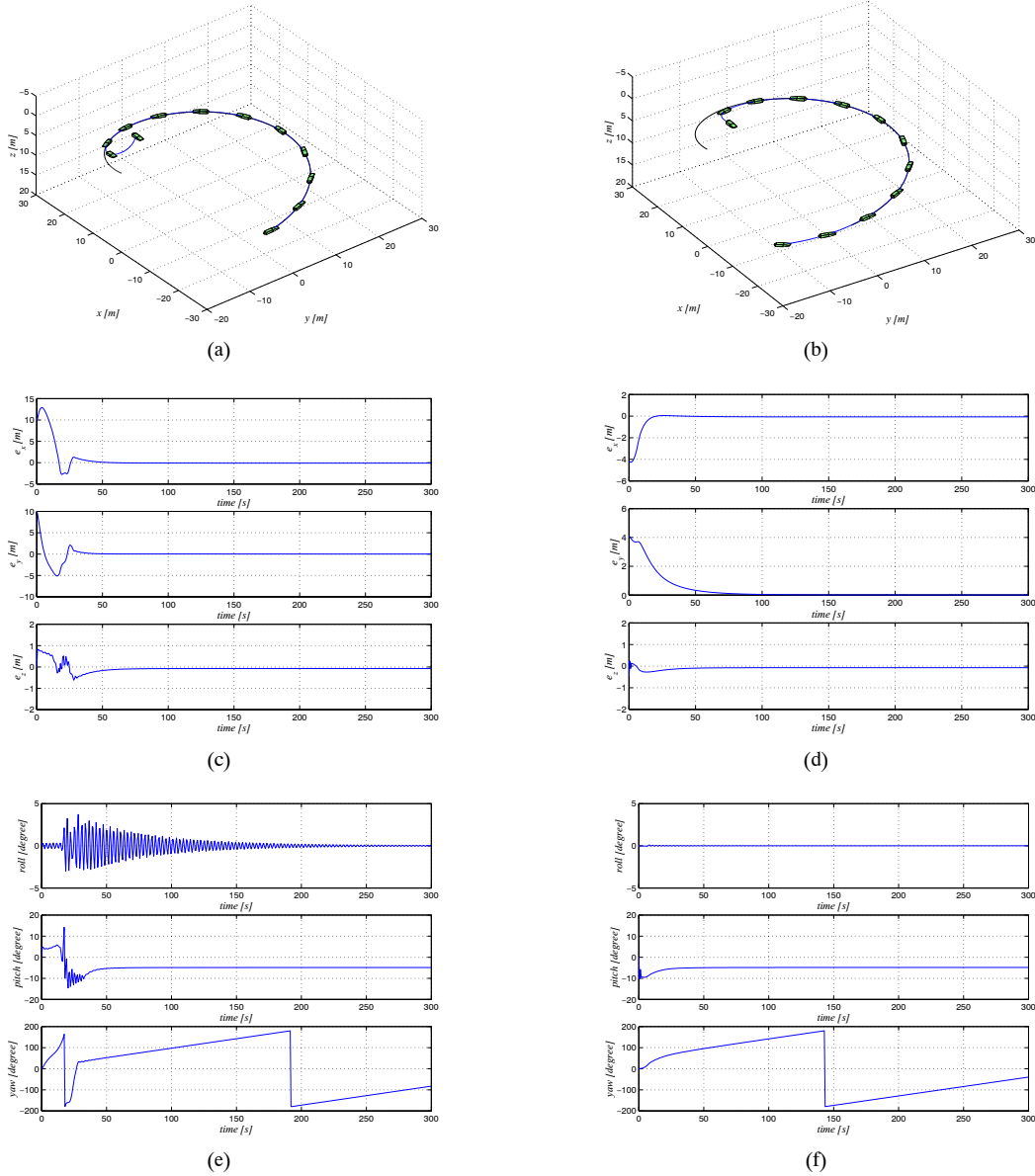


Fig. 4. Vehicle trajectory in 3-D space using the trajectory-tracking controller presented in Section III (diagram (a)), and the path-following controller (diagram (b)). Time evolution of the position error $e = (e_x, e_y, e_z)$, the roll ϕ , pitch θ , and yaw ψ Euler angles for the trajectory-tracking (diagrams (c), (e)), and the path-following (diagrams (d), (f)).

where λ_μ is any positive constant that satisfies

$$\lambda_\mu < \min\{\lambda_{\min}(L_v), \lambda_{\min}(L_\omega)\}. \quad (42)$$

Computing the time derivative of V_v and V_ω along the solutions of (2b), (3), and (16) for $\Theta = \Theta^*$, we obtain

$$\begin{aligned} \dot{V}_v &= -e^{2\lambda_\mu t} [\tilde{v}' L_v \tilde{v} - \lambda_\mu \|\tilde{v}\|^2] \\ &\quad - e^{2\lambda_\mu t} \beta_{v_{\Theta^*}}(\hat{v}, \hat{\omega}, v, \omega, p, R) \|\tilde{v}\|^2 \end{aligned} \quad (43a)$$

$$\begin{aligned} \dot{V}_\omega &= -e^{2\lambda_\mu t} [\tilde{\omega}' L_\omega \tilde{\omega} - \lambda_\mu \|\tilde{\omega}\|^2] \\ &\quad - e^{2\lambda_\mu t} \beta_{\omega_{\Theta^*}}(\hat{v}, \hat{\omega}, \omega, p, R) \|\tilde{\omega}\|^2. \end{aligned} \quad (43b)$$

To prove (18a), consider the Lyapunov function $V := \frac{1}{2} \|\mathbf{e}_{\Theta^*}\|^2$, which is the same as $V = V_v + V_\omega$ with $\lambda_\mu = 0$. From (43) it follows that $\dot{V} \leq -2 \min\{\lambda_{\min}(L_v), \lambda_{\min}(L_\omega)\} V$. Using the Comparison

Lemma [45] one concludes that (18a) holds with $\kappa = \min\{\lambda_{\min}(L_v), \lambda_{\min}(L_\omega)\}$ and $\gamma_1 = \|\mathbf{e}_{\Theta^*}(0)\|$.

To prove (18b), observe from (43a) and (42) that

$$\begin{aligned} \dot{V}_v &\leq -e^{2\lambda_\mu t} \beta_{v_{\Theta^*}}(\hat{v}, \hat{\omega}, v, \omega, p, R) \|\tilde{v}\|^2 \\ &\leq -2\beta_{v_{\Theta^*}}(\hat{v}, \hat{\omega}, v, \omega, p, R) V_v. \end{aligned}$$

Defining $W_v := \sqrt{V_v} = \frac{1}{\sqrt{2}} e^{\lambda_\mu t} \|\tilde{v}\|$ and taking an interval $[0, \tau)$ on which $V_v > 0$, we have $\dot{W}_v \leq -\beta_{v_{\Theta^*}}(\hat{v}, \hat{\omega}, v, \omega, p, R) W_v$, $t \in [0, \tau)$ and therefore

$$W_v(t) - W_v(0) \leq - \int_0^t \beta_{v_{\Theta^*}}(\hat{v}(\tau), \hat{\omega}(\tau), v(\tau), \omega(\tau), p(\tau), R(\tau)) W_v(\tau) d\tau, \quad t \in [0, \tau)$$

Consequently,

$$\int_0^t e^{\lambda\mu\tau} \beta_{v_{\Theta^*}}(\hat{v}(\tau), \hat{\omega}(\tau), v(\tau), \omega(\tau), p(\tau), R(\tau)) \|\tilde{v}(\tau)\| d\tau \leq \gamma_2 < \infty, \quad t \in [0, \tau]$$

with $\gamma_2 = \|\tilde{v}(0)\|$. On the other hand, if $V_v(\tau) = 0$ for some $\tau > 0$, $V_v(t) = 0$ for all $t \geq \tau$. Therefore,

$$\begin{aligned} & \int_0^t e^{\lambda\mu\tau} \beta_{v_{\Theta^*}}(\hat{v}(\tau), \hat{\omega}(\tau)v(\tau), \omega(\tau), p(\tau), R(\tau)) \|\tilde{v}(\tau)\| d\tau \\ &= \int_0^\tau e^{\lambda\mu\tau} \beta_{v_{\Theta^*}}(\hat{v}(\tau), \hat{\omega}(\tau), v(\tau), \omega(\tau), p(\tau), R(\tau)) \|\tilde{v}(\tau)\| d\tau \leq \gamma_2 < \infty \end{aligned}$$

Inequality (18c) can be also concluded by applying the same arguments to (43b). In that case $\gamma_3 = \|\tilde{\omega}(0)\|$. ■

Lemma 2

Proof: Taking norms to $h_{3\Theta}(\cdot)$ defined in (20) and using (17a) we conclude that

$$\begin{aligned} \|h_{3\Theta}(\cdot)\| &\leq [\|\mathbf{M}_{\Theta}^2 L_v\|_2 + k_{e_{\Theta}} \lambda_{max}(\mathbf{M}_{\Theta})] \|\tilde{v}_{\Theta}\| \\ &+ \lambda_{max}(\mathbf{M}_{\Theta}) \beta_{v_{\Theta}}(\cdot) [\lambda_{max}(\mathbf{M}_{\Theta}) + c_1] \|\tilde{v}_{\Theta}\| \\ &+ \lambda_{max}(\mathbf{M}_{\Theta}) [\|\mathbf{M}_{\Theta} \delta_{\Theta}\| \\ &+ 2\lambda_{max}(\mathbf{M}_{\Theta}) \|\dot{p}_d\|] \|\tilde{\omega}_{\Theta}\| \end{aligned}$$

Also, from (17b)–(17c), a bound for $h_{4\Theta}(\cdot)$ given in (27) can be computed as follows.

$$\begin{aligned} \|h_{4\Theta}(\cdot)\| &\leq \|S(\hat{v}_{\Theta})\mathbf{M}_{\Theta}\hat{v}_{\Theta} - S(\hat{v}_{\Theta})\mathbf{M}_{\Theta}v + S(\hat{v}_{\Theta})\mathbf{M}_{\Theta}v \\ &- S(v)\mathbf{M}_{\Theta}v\| + \|S(\hat{\omega}_{\Theta})\mathbf{J}_{\Theta}\hat{\omega}_{\Theta} - S(\hat{\omega}_{\Theta})\mathbf{J}_{\Theta}\omega \\ &+ S(\hat{\omega}_{\Theta})\mathbf{J}_{\Theta}\omega - S(\omega)\mathbf{J}_{\Theta}\omega\| + \|f_{\omega_{\Theta}}(v, \omega, p, R) \\ &- f_{\omega_{\Theta}}(\hat{v}_{\Theta}, \omega, p, R) + f_{\omega_{\Theta}}(\hat{v}_{\Theta}, \omega, p, R) \\ &- f_{\omega_{\Theta}}(\hat{v}_{\Theta}, \hat{\omega}_{\Theta}, p, R)\| + [\|\mathbf{J}_{\Theta}L_{\omega}\|_2 \\ &+ \beta_{\omega_{\Theta}}(\cdot)\lambda_{max}(\mathbf{J}_{\Theta})\|\tilde{\omega}_{\Theta}\| + \|[0_{3 \times 1} \mathbf{J}_{\Theta}] \varphi_{v_{\Theta}}(\cdot)\| \\ &+ \|[0_{3 \times 1} \mathbf{J}_{\Theta}] \varphi_{\omega_{\Theta}}(\cdot)\|] \\ &\leq c_2 \beta_{2\Theta}(\hat{v}_{\Theta}, \hat{\omega}_{\Theta}, v, \omega, p, R) \|\tilde{v}_{\Theta}\| + [(\lambda_{max}(\mathbf{J}_{\Theta}) \\ &+ c_3)\beta_{\omega_{\Theta}}(\hat{v}_{\Theta}, \hat{\omega}_{\Theta}, \omega, p, R) + \|\mathbf{J}_{\Theta}L_{\omega}\|_2] \|\tilde{\omega}_{\Theta}\| \end{aligned}$$

Using these two bounds in (26) and resorting to Young's inequality it follows that for every positive constant γ_i , $i = 1, \dots, 9$, \dot{V}_3 satisfies

$$\begin{aligned} \dot{V}_3 &\leq -\left[k_{e_{\Theta}} \lambda_{min}(\mathbf{M}_{\Theta}^{-1}) - \frac{\gamma_1}{2} - \frac{\gamma_2}{2}\right] \|e\|^2 \\ &- \left[\lambda_{min}(K_{\varphi_{\Theta}}) - \frac{\gamma_3}{2} \|\mathbf{M}_{\Theta}^2 L_v\|_2 + k_{e_{\Theta}} \lambda_{max}(\mathbf{M}_{\Theta})\right] \\ &- \frac{\gamma_4}{2} \lambda_{max}(\mathbf{M}_{\Theta}) \beta_{v_{\Theta}}(\cdot) (\lambda_{max}(\mathbf{M}_{\Theta}) + c_1) \|\tilde{v}_{\Theta}\| \\ &- \frac{\gamma_5}{2} \lambda_{max}(\mathbf{M}_{\Theta}) [\|\mathbf{M}_{\Theta} \delta_{\Theta}\| + 2\lambda_{max}(\mathbf{M}_{\Theta}) \|\dot{p}_d\|] \|\varphi_{\Theta}\|^2 \\ &+ \left[\frac{1}{2\gamma_2} + \frac{1}{2\gamma_3} [\|\mathbf{M}_{\Theta}^2 L_v\|_2 + k_{e_{\Theta}} \lambda_{max}(\mathbf{M}_{\Theta})]\right] \|\tilde{v}_{\Theta}\|^2 \\ &+ \left[\frac{1}{2\gamma_5} \lambda_{max}(\mathbf{M}_{\Theta}) (\|\mathbf{M}_{\Theta} \delta_{\Theta}\| + 2\lambda_{max}(\mathbf{M}_{\Theta}) \|\dot{p}_d\|)\right] \\ &+ \frac{1}{2\gamma_9} \|\mathbf{J}_{\Theta}L_{\omega}\|_2 \|\tilde{\omega}_{\Theta}\|^2 + \left[\frac{1}{2\gamma_4} \lambda_{max}(\mathbf{M}_{\Theta}) \beta_{v_{\Theta}}(\cdot)\right] \end{aligned}$$

$$\begin{aligned} &\times (\lambda_{max}(\mathbf{M}_{\Theta}) + c_1) + \frac{1}{2\gamma_7} c_2 \beta_{2\Theta}(\cdot) \|\tilde{v}_{\Theta}\| \\ &+ \frac{1}{2\gamma_8} (\lambda_{max}(\mathbf{J}) + c_3) \beta_{\omega_{\Theta}}(\cdot) \|\tilde{\omega}_{\Theta}\| \\ &- \left[\lambda_{min}(K_{z_{2\Theta}}) - \frac{\gamma_7}{2} c_2 \beta_{2\Theta}(\cdot) \|\tilde{v}_{\Theta}\| - \frac{\gamma_8}{2} (\lambda_{max}(\mathbf{J}_{\Theta})\right. \\ &\left. + c_3) \beta_{\omega_{\Theta}}(\cdot) \|\tilde{\omega}_{\Theta}\| - \frac{\gamma_9}{2} \|\mathbf{J}_{\Theta}L_{\omega}\|_2\right] \|z_{2\Theta}\|^2 + \frac{1}{2\gamma_1} \|\delta_{\Theta}\|^2 \end{aligned}$$

where we used the facts that $\|\tilde{v}_{\Theta}\| \|\varphi_{\Theta}\| \leq \|\tilde{v}_{\Theta}\| \left[\frac{1}{2\gamma_4} + \frac{\gamma_4}{2} \|\varphi\|^2\right]$ and $\|\tilde{\omega}_{\Theta}\| \|z_{2\Theta}\| \leq \|\tilde{\omega}_{\Theta}\| \left[\frac{1}{2\gamma_8} + \frac{\gamma_8}{2} \|z_{2\Theta}\|^2\right]$. Therefore, there exist sufficiently small positive constants λ , γ_i , $i = 1, \dots, 9$, and sufficiently large positive constants κ_j , $j = 1, \dots, 5$ such that

$$\begin{aligned} \dot{V}_3 &\leq -\lambda V_3 + [\kappa_1 \beta_{v_{\Theta}}(\cdot) \|\tilde{v}_{\Theta}\| + \kappa_2 \beta_{\omega_{\Theta}}(\cdot) \|\tilde{\omega}_{\Theta}\|] V_3 \\ &+ \kappa_3 \|\mathbf{e}_{\Theta}\|^2 + \kappa_4 \beta_{v_{\Theta}}(\cdot) \|\tilde{v}_{\Theta}\| + \kappa_5 \beta_{\omega_{\Theta}}(\cdot) \|\tilde{\omega}_{\Theta}\| \\ &+ \frac{1}{2\gamma_1} \|\delta_{\Theta}\|^2 \end{aligned} \quad (44)$$

where in view of (29)–(30), the signals $\kappa_1 \beta_{1\Theta}(\cdot) \|\tilde{v}_{\Theta}\|$, $\kappa_2 \beta_{\omega_{\Theta}}(\cdot) \|\tilde{\omega}_{\Theta}\|$, $\kappa_4 \beta_{v_{\Theta}}(\cdot) \|\tilde{v}_{\Theta}\|$, $\kappa_5 \beta_{\omega_{\Theta}}(\cdot) \|\tilde{\omega}_{\Theta}\|$ are \mathcal{L}_1 defined on $[T^\dagger, T)$, and $\kappa_3 \|\mathbf{e}_{\Theta}\|^2$ is \mathcal{L}_2 defined on $[T^\dagger, T)$. Applying the Comparison Lemma to (44) and using [46, Lemma 1] we conclude that V_3 is bounded on $[T^\dagger, T)$. Moreover, when $T = +\infty$, V_3 converges to a ball of radius $\frac{\|\delta_{\Theta}\|^2}{2\lambda\gamma_1}$ as $t \rightarrow \infty$. Standard signal chasing arguments can now be applied to conclude that the signals $\hat{v}_{\Theta}(t)$, $\hat{\omega}_{\Theta}(t)$, $u_{\Theta}(t)$ and $e(t)$ in closed-loop system remain bounded. Furthermore, applying the same arguments described in the proof of item ii) of Theorem 1, one concludes that the tracking error $\|p(t) - p_d(t)\|$ converges to a neighborhood of the origin that can be made arbitrarily small. ■

Derivation of (35)

Using (16) for $\Theta = \Theta^*$ and the fact that $v = \hat{v}_{\Theta^*} - \tilde{v}_{\Theta^*} = \hat{v}_{\Theta^\dagger} - \tilde{v}_{\Theta^\dagger}$, and $\omega = \hat{\omega}_{\Theta^*} - \tilde{\omega}_{\Theta^*} = \hat{\omega}_{\Theta^\dagger} - \tilde{\omega}_{\Theta^\dagger}$, the time-derivative of V in (34) can be written as

$$\begin{aligned} \dot{V} &= -\hat{v}'_{\Theta^*} \mathbf{M} L_v \hat{v}_{\Theta^*} - \hat{\omega}'_{\Theta^*} \mathbf{J} L_{\omega} \hat{\omega}_{\Theta^*} \\ &+ \hat{v}'_{\Theta^*} [-S(\hat{\omega}_{\Theta^\dagger} - \tilde{\omega}_{\Theta^\dagger}) \mathbf{M} \hat{v}_{\Theta^\dagger} + S(\omega) \mathbf{M} \tilde{v}_{\Theta^\dagger} \\ &+ f_v(\hat{v}_{\Theta^\dagger}, p, R) + f_v(v, p, R) - f_v(\hat{v}_{\Theta^\dagger}, p, R) \\ &+ g_v [1 \ 0_{1 \times 3}] \mathcal{K}_{\Theta^\dagger}(\cdot) + \mathbf{M} L_v (\hat{v}_{\Theta^\dagger} - \tilde{v}_{\Theta^\dagger}) \\ &- \beta_{v_{\Theta^*}}(\hat{v}_{\Theta^*}, \hat{\omega}_{\Theta^*}, v, \omega, p, R) \mathbf{M} \tilde{v}_{\Theta^*}] \\ &+ \hat{\omega}'_{\Theta^*} [-S(\hat{v}_{\Theta^\dagger} - \tilde{v}_{\Theta^\dagger}) \mathbf{M} (\hat{v}_{\Theta^\dagger} - \tilde{v}_{\Theta^\dagger}) - S(\hat{\omega}_{\Theta^\dagger} \\ &- \tilde{\omega}_{\Theta^\dagger}) \mathbf{J} (\hat{\omega}_{\Theta^\dagger} - \tilde{\omega}_{\Theta^\dagger}) + f_{\omega}(\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, p, R) \\ &+ f_{\omega}(v, \omega, p, R) - f_{\omega}(\hat{v}_{\Theta^\dagger}, \omega, p, R) + f_{\omega}(\hat{v}_{\Theta^\dagger}, \omega, p, R) \\ &- f_{\omega}(\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, p, R) + G_{\omega} [0_{3 \times 1} \ I_{3 \times 3}] \mathcal{K}_{\Theta^\dagger}(\cdot) \\ &- \mathbf{J} L_{\omega} (\hat{\omega}_{\Theta^*} - \tilde{\omega}_{\Theta^*}) - \beta_{\omega_{\Theta^*}}(\hat{v}_{\Theta^*}, \hat{\omega}_{\Theta^*}, \omega, p, R) \mathbf{J} \tilde{\omega}_{\Theta^*}]. \end{aligned}$$

Using the Young's inequality, after a straightforward but messy computations, one can conclude that there exist a sufficiently small $\lambda > 0$ and sufficiently large $\kappa_j > 0$, $j = 1, \dots, 13$ such that for all $t \in [T^\dagger, T)$, inequality (35) holds with

$$\begin{aligned} a_1(t) &= \kappa_2 \beta_{1\Theta^\dagger}(\hat{v}_{\Theta^\dagger}, v, \omega, p, R) \|\tilde{v}_{\Theta^\dagger}\| + \kappa_3 \beta_{v_{\Theta^*}}(\hat{v}_{\Theta^*}, \hat{\omega}_{\Theta^*}, \\ &v, \omega, p, R) \|\tilde{v}_{\Theta^*}\| + \kappa_4 \beta_{2\Theta^\dagger}(\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, v, \omega, p, R) \|\tilde{v}_{\Theta^\dagger}\| \end{aligned}$$

$$\begin{aligned}
& +\kappa_5\beta_{\omega_{\Theta^\dagger}}(\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, \omega, p, R)\|\tilde{\omega}_{\Theta^\dagger}\| \\
& +\kappa_6\beta_{\omega_{\Theta^*}}(\hat{v}_{\Theta^*}, \hat{\omega}_{\Theta^*}, v, \omega, p, R)\|\tilde{\omega}_{\Theta^*}\| \\
a_2(t) & = \kappa_7\|\tilde{e}_{\Theta^\dagger}\|^2 \\
b_1(t) & = \kappa_8\beta_{1_{\Theta^\dagger}}(\hat{v}_{\Theta^\dagger}, v, \omega, p, R)\|\tilde{v}_{\Theta^\dagger}\| + \kappa_9\beta_{v_{\Theta^*}}(\hat{v}_{\Theta^*}, \hat{\omega}_{\Theta^*}, \\
& v, \omega, p, R)\|\tilde{v}_{\Theta^*}\| + \kappa_{10}\beta_{2_{\Theta^\dagger}}(\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, v, \omega, p, R)\|\tilde{v}_{\Theta^\dagger}\| \\
& +\kappa_{11}\beta_{\omega_{\Theta^\dagger}}(\hat{v}_{\Theta^\dagger}, \hat{\omega}_{\Theta^\dagger}, \omega, p, R)\|\tilde{\omega}_{\Theta^\dagger}\| \\
& +\kappa_{12}\beta_{\omega_{\Theta^*}}(\hat{v}_{\Theta^*}, \hat{\omega}_{\Theta^*}, v, \omega, p, R)\|\tilde{\omega}_{\Theta^*}\|, \\
b_2(t) & = \kappa_{13}\|\tilde{e}_{\Theta^\dagger}\|^2.
\end{aligned}$$

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