Nonlinear Control Systems

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6. Nonlinear Design

IST-DEEC PhD Course

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2012

Consider the following system

$$\begin{aligned} \dot{x}_1 &= x_2\\ \dot{x}_2 &= h(x) + g(x)u \end{aligned}$$

where h, g are unknown nonlinear functions and $g(x) \ge g_0 > 0, \forall x$.

Goal: Design a state-feedback control law to stabilize the origin.

Idea: Design a control law that restrict the motion of the system to the manifold or surface

$$s = a_1 x_1 + x_2 = 0, \quad a_1 > 0$$

Note that the motion on the manifold s = 0 satisfies

$$x_2 = -a_1 x_1 \longrightarrow \dot{x}_1 = -a x_1 \longrightarrow x = (x_1, x_2) \rightarrow 0$$

and furthermore the motion is independent of h and g!

Now the question is how can we bring the trajectory to the manifold?

Let

$$V = \frac{1}{2}s^2$$

and therefore

$$\begin{split} \dot{V} &= s\dot{s} \\ &= s(a_1\dot{x}_1 + \dot{x}_2) \\ &= s(a_1x_2 + h(x)) + sg(x)u \end{split}$$

Suppose that $\left|\frac{a_1x_2+h(x)}{g(x)}\right| \leq \rho(x), \ \forall x \in \mathbb{R}^2$ and assume that $\rho(x)$ is known. Then,

$$\dot{V} \leq |s|
ho(x)g(x) + su \ = g(x)|s|[
ho(x) + \mathrm{sgn}(s)u]$$

Let $u = -\beta(x)\operatorname{sgn}(s)$ with $\beta(x) \ge \rho(x) + \beta_0$, $\beta_0 > 0$. $\dot{V} < -q(x)|s|\beta_0 < -q_0\beta_0|s|$

Let
$$W = \sqrt{2V} = |s|$$
 (Note that $\sqrt{u}' = \frac{u'}{2\sqrt{u}}$)

The upper right-hand derivative is given by

$$D^+W = \frac{2\dot{V}}{2\sqrt{2V}} = \frac{\dot{V}}{W} \le -g_0\beta_0\frac{W}{W}$$

By the comparison lemma

$$W(s(t)) \le W(s(0)) - g_0 \beta_0 t$$

Thus, the trajectory reaches the manifold s=0 in finite time. Moreover, once it reaches the manifold $\dot{V}\leq -g_0\beta_0|s|=0$, which means that it cannot leave from it.

In summary, for the example above, the sliding mode control strategy is composed by two phases:

- 1. reaching phase: the trajectory starting off the manifold $s=0\ {\rm move}$ toward it and reach it in finite time.
- 2. sliding phase: the motion is confined to the manifold s = 0 and the dynamics of the system are represented by the reduced-order model $\dot{x}_1 = -a_1x_1$.

Remark: The control law $u = -\beta(x)\text{sgn}(s)$ is called a sliding mode control law. Note that it is robust with respect to uncertainty on h and g. We only need to know the upper form $\rho(x)$.

Furthermore, if $\left|\frac{a_1x_2+h(x)}{g(x)}\right| \leq k_1, \forall x \in D$ then $u = -k \operatorname{sgn}(s), \ k > k_1$ and if k can be chosen arbitrarily large, it can achieve semi-global stability.

However, due to imperfections in switching devices and delays, sliding mode control suffers from chattering!

What are the consequences of this zig-zag motion (oscillation)?

- High heat losses.
- High wear of moving mechanical parts.
- · It may excite unmodelled high frequency dynamics
- Degrades performance and may lead to instability

To eliminate chattering replace u by

$$u = -\beta(x)\mathsf{sat}(s/\epsilon)$$

where

$$\mathsf{sat}(y) = \left\{ \begin{array}{ll} y, & \text{if } |y| \leq 1 \\ \mathsf{sgn}(y), & \text{otherwise} \end{array} \right.$$

In that case we have,

$$\dot{V} \le -g_0 \beta_0 |s|,$$

while $|s| \ge \epsilon$, which means that it reaches in finite time the set $\{|s| \le \epsilon\}$.

Inside the boundary layer $|s| = \epsilon$, we have

$$s = a_1 x_1 + x_2 = a_1 x_1 + \dot{x}_1 \quad \longrightarrow \quad \dot{x}_1 = -a_1 x_1 + s$$

Thus, let

$$V_1 = \frac{1}{2}x_1^2$$

Then,

$$\begin{split} \dot{V} &\leq -a_1 x_1^2 + x_1 s \\ &\leq -a_1 x_1^2 + |x_1| \epsilon \\ &= -(1-\theta) a_1 x_1^2 - \theta a_1 x_1^2 + |x_1| \epsilon \\ &\leq -(1-\theta) a_1 x_1^2, \quad \forall |x_1| \geq \frac{\epsilon}{\theta a_1}, \ 0 < \theta < 1 \end{split}$$

Thus the trajectory reaches the set

$$\Omega_{\epsilon} = \left\{ |x_1| \le \frac{\epsilon}{a_1 \theta}, \ |s| \le \epsilon \right\}$$

in finite time, and therefore we can also conclude that it is ultimately bounded.

Stabilization

Consider the system

$$\dot{x} = f(x) + B(x)[G(x)E(x)u + \delta(t, x, u)]$$

where $x \in \mathbb{R}^n$ is the state, $u \in \mathbb{R}^p$ is the input, f, B, G and E are sufficiently smooth functions, and G and δ are unknown (uncertainties). Consider also that G is diagonal and positive definite with $g_i(x) \ge g_0 > 0$, E(x) is nonsingular, f(0) = 0, and x = 0 is on open-loop equilibrium point (with $\delta = 0$).

Suppose that there is a diffeomorphic coordinate transformation $\begin{bmatrix} \eta \\ \xi \end{bmatrix} = T(x)$ (that is, $\frac{\partial T}{\partial x}$ is nonsingular, and T is proper, i.e, $\lim_{\|x\|\to\infty} \|T(x)\| = \infty$), with $\eta \in \mathbb{R}^{n-p}$, $\xi \in \mathbb{R}^p$ such that

$$\frac{\partial T}{\partial x}B(x) = \left[\begin{array}{c} 0\\ I \end{array}\right]$$

Then

$$\left[\begin{array}{c} \dot{\eta}\\ \dot{\xi}\end{array}\right]=\frac{\partial T}{\partial x}f(x)+\frac{\partial T}{\partial x}B(x)[G(x)E(x)u+\delta(t,x,u)]$$

and so we obtain the system written in the so-called regular form:

$$\dot{\eta} = f_a(\eta, \xi) \tag{1}$$

$$\dot{\xi} = f_b(\eta, \xi) + G(x)E(x)u + \delta(t, x, u)$$
(2)

Step 1

Design the sliding manifold $s = \xi - \phi(\eta) = 0$ to stabilize (1), that is, when the motion is restricted to the manifold the reduced-order model

 $\dot{\eta} = f_a(\eta, \phi(\eta))$

has an asymptotically stable equilibrium point at the origin.

This is the same as to solve the stabilization problem for the system

$$\dot{\eta} = f_a(\eta, \xi)$$

with ξ viewed as the control input.

Consider also that $\phi(\eta)$ is designed such that the system $\dot{\eta} = f_a(\eta, \phi(\eta) + s)$ is local ISS when s is viewed as the input.

Step 2

Design the control u to bring s to the boundary layer $\{|s_i| \leq \epsilon, 1 \leq i \leq p\}$ in finite time and keep it there $\forall t \geq T \geq 0.$

$$\dot{s} = \dot{\xi} - \frac{\partial \phi}{\partial \eta} \dot{\eta} = f_b(\eta, \xi) + G(x)E(x)u + \delta(t, x, u) - \frac{\partial \phi}{\partial \eta}f_a(\eta, \xi)$$

Let

$$u = E^{-1}(x)\hat{G}^{-1}(x)[f_b(\eta,\xi) - \frac{\partial\phi}{\partial\eta}f_a(\eta,\xi)] + E^{-1}(x)v$$

Then,

$$\dot{s}_i = g_i(x)v_i + \Delta_i(t, x, v), \quad i = 1, ..., p$$

where $\Delta_i(t, x, v)$ is the *i*th component of $\Delta(t, x, v) = [I - G(x)\hat{G}^{-1}(x)][f_b(.) - \frac{\partial\phi}{\partial\eta}f_a(.)] + \delta(t, x, u)$ Assume that

$$\left|\frac{\Delta_i(t, x, u)}{g_i(x)}\right| \le \rho(x) + k_0 \|v\|_{\infty}$$

with $k_0 \in [0, 1]$.

Then

$$V_i = \frac{1}{2}s_i^2$$

$$\begin{split} \dot{V}_i &= s_i \dot{s}_i \\ &= s_i g_i(x) v_i + s_i \Delta_i(t, x, u) \\ &\leq |s_i| g_i(x) [v_i \text{sgn}(s_i) + \rho(x) + k_0 \|v\|_\infty] \end{split}$$

Take $v_i = -\beta(x) \operatorname{sat}(\frac{s_i}{\epsilon})$

$$\dot{V}_i \leq |s_i|g_i(x) \left[-\beta(x) \mathsf{sat}\Big(\frac{|s_i|}{\epsilon}\Big) + \rho(x) + k_0\beta(x) \right]$$

In the region $|s_i| \geq \epsilon$ we have

$$\begin{split} \dot{V}_i &\leq |s_i|g_i(x) \left[-(1-k_0)\beta(x) + \rho(x) \right] \\ &\leq -g_0\beta_0(1-k_0)|s_i| \end{split}$$

by setting $\beta(x) \ge \frac{\rho(x)}{1-k_0} + \beta_0$, $\beta_0 > 0$.

Thus, $|s_i(t)|$ will decrease until it reaches the set $\{|s_i| \leq \epsilon\}$ in finite time and remains inside thereafter.

We can conclude that the sliding mode controller achieves ultimate boundedness with an ultimate bound that can be controlled by the design parameter ϵ . Moreover it is robust with respect to matched uncertainties.

Example:

$$\dot{x}_1 = x_2 + \theta_1 x_1 \sin x_2$$

 $\dot{x}_2 = \theta_2 x_2^2 + x_1 + u$

where θ_1, θ_2 are unknown parameters and $|\theta_1| \leq a$ and $|\theta_2| \leq b$ with a, b known. Note that the system is already in the regular form.

Step 1

Design
$$x_2$$
 to robustly stabilize $x_1=0$
$$\dot{x}_1=x_2+\theta_1 x_1 \sin x_2$$

$$V=\frac{1}{2}x_1^2$$

$$\dot{V} = x_1 x_2 + \theta_1 x_1^2 \sin x_2$$

Let $x_2 = -kx_1$. Then

$$\begin{split} \dot{V} &= -kx_1^2 + \theta_1 x_1^2 \sin(-kx_1) \\ &\leq -kx_1^2 + \theta_1 x_1^2 \\ &\leq -(k-a)x_1^2 \end{split}$$

for k > a. Thus, the sliding manifold is

$$s = x_2 + kx_1 = 0$$

Step 2

$$V=\frac{1}{2}s^2$$

$$\dot{V} = s[\theta_2 x_2^2 + x_1 + u + k(x_2 + \theta_1 x_1 \sin x_1)]$$

Let

$$\begin{split} u &= -x_1 - kx_2 + v \\ \dot{V} &= s[\theta_2 x_2^2 + k \theta_1 x_1 \sin x_2 + v] \end{split}$$
 where $|\Delta(x)| &= |\theta_2 x_2^2 + k \theta_1 x_1 \sin x_2| \leq a k |x_1| + b x_2^2.$

 $\begin{array}{l} {\rm Choose}\;v=-\beta(x){\rm sat}(\frac{s}{\epsilon})\;{\rm with}\;\beta(x)=ak|x_1|+bx_2^2+\beta_0,\;\beta_0>0.\\ {\rm Then\;for\;}|s|\geq\epsilon\end{array}$

$$\begin{split} \dot{V} &\leq |s| \left(|\Delta(x)| - \beta(x) \mathsf{sat}(\frac{|s|}{\epsilon}) \right) \\ &\leq -\beta_0 |s| \end{split}$$

so |s| reaches in finite time the boundary layer $\{|s| \leq \epsilon\}$.

Example: Tracking

Consider the SISO system

$$\begin{split} \dot{x} &= f(x) + g(x)[u + \delta(t, x, u)] \\ y &= h(x) \end{split}$$

Normal form

$$\begin{split} \dot{\eta} &= f_0(\eta, \xi) \\ \dot{\xi}_1 &= \xi_2 \\ \dot{\xi}_2 &= \xi_3 \\ \vdots \\ \dot{\xi}_{\rho-1} &= \xi_{\rho} \\ \dot{\xi}_{\rho} &= a(x) + b(x)[u + \delta(t, x, u)] \\ y &= \xi_1 \end{split}$$

Suppose that $\dot{\eta} = f_0(\eta, \xi)$ is ISS with ξ as input. <u>Goal</u>: Track the reference r(t) and suppose that $\dot{r}, \ \ddot{r}, ..., r^{(\rho)}$ are available.

Define

$$e_1 = \xi_1 - r$$

$$e_2 = \xi_2 - \dot{r}$$

$$\vdots$$

$$e_\rho = \xi_\rho - r^{\rho - 1}$$

Then

$$\dot{\eta} = f_0(\eta, \xi) \tag{3}$$

$$\dot{e}_1 = e_2 \tag{4}$$

$$\dot{e}_{\rho-1} = e_{\rho} \tag{5}$$

$$\dot{e}_{\rho} = a(x) + b(x)[u + \delta(t, x, u)] - r^{(\rho)}(t)$$
 (6)

Note that (3) is ISS and (4)-(5) is as a linear system (written in the controllable canonical form) with e_{ρ} viewed as input.

Therefore, for the linear subsystem select the linear control law

$$e_{\rho} = -(k_1e_1 + k_2e_2 + \dots + k_{\rho-1}e_{\rho-1})$$

where k_1 to $k_{\rho-1}$ are chosen such that the closed-loop system is Hurwitz. Then, the sliding manifold is

$$s = k_1 e_1 + \dots + k_{\rho-1} e_{\rho-1} + e_{\rho} = 0$$

Note that for $\rho = 2$, we have $s = k_1 e_1 + e_2 = k_1 e_1 + \dot{e}_1 = 0$.

To converge to the sliding manifold, choose

$$V = \frac{1}{2}s^2$$

Then,

$$\dot{V} = s \left(k_1 e_1 + \dots + k_{\rho-1} e_{\rho-1} + e_{\rho} + a(x) + b(x) [u + \delta(t, x, u)] - r^{(\rho)} \right)$$

Let

$$u = -\frac{1}{b(x)} \left(k_2 e_2 + \dots + k_{\rho-1} e_{\rho} - r^{(\rho)} + v \right)$$

$$\begin{split} \dot{V} &= s\left(v + b(x)\delta(t,x,-\frac{1}{b(x)}[k_1e_2....r^{(\rho)}] + v)\right)\\ \text{Defining }\Delta(t,x,v) &= b(x)\delta(t,x,-\frac{1}{b(x)}[k_1e_2....r^{(\rho)}] + v\\ \text{if }|\Delta(t,x,v)| &\leq \rho(x) + k_0|v|, \text{ with } k_0 \in [0,1). \text{ Then, selecting} \end{split}$$

$$v = -\beta(x)\mathsf{sat}(rac{s}{\epsilon}), \ \beta(x) \ge rac{
ho(x)}{1-k_0} + \beta_0, \ \beta_0 > 0$$

we can conclude that there exists a finite time $T \ge t_0$ such that the tracking error |y(t) - r(t)| will be trapped inside a small neighborhood (that depends on ϵ) for all $t \ge T$.

Sliding mode with integral control

If the reference signal r(t) = r is constant, we can achieve zero steady-state error using integral control.

To this effect define $e_0(t) = \int_0^t (y(\tau) - r) d\tau \longrightarrow \dot{e}_0 = y - r.$ Then we have

> $\dot{\eta} = f_0(\eta, \xi)$ $\dot{e}_0 = e_1$ $\dot{e}_1 = e_2$ \vdots $\dot{e}_{\rho-1} = e_{\rho}$ $\dot{e}_{\rho} = a(x) + b(x)[u + \delta(\cdot)]$

and

$$s = k_0 e_0 + k_1 e_1 + \ldots + k_{\rho-1} e_{\rho-1} + e_{\rho}$$

In fact, for $\beta(x) = k$ and $v = k \operatorname{sat}(\frac{s}{\epsilon})$ we have that for $\rho = 1$ (relative degree one) the control algorithm turns out to be a classical Proportional Integral (PI) + saturation feedback law.

If the relative degree is two ($\rho=2)$ then we obtain a PID (Proportional, Integral and Derivative) structure + saturation

Nonlinear Lyapunov based control

 $\frac{\textbf{Example:}}{(\text{AUV})} \text{ Pose stabilization of a fully actuated Autonomous Underwater Vehicle}$

Consider the model of a fully actuated AUV

$$\begin{split} M \, \dot{\nu} + C(\nu) \, \nu + D(\nu) \, \nu + g(\eta) &= \tau \\ \dot{\eta} &= J(\eta) \, \nu \end{split}$$

where $\tau \in \mathbb{R}^6$ is the control input (forces and torques), $\eta \in \mathbb{R}^6$ is the position and orientation, $\nu \in \mathbb{R}^6$ is the linear and angular velocities, $M = M^T > 0$ is the rigid body and added mass inertia matrix, $C(\nu) = -C(\nu)^T$ is the matrix of Coriolis and Centrifugal terms, $D(\nu) > 0$ is the damping matrix, and $g(\eta)$ is the restoring term (buoyancy and gravity).

Goal: Design a state feedback control so that $\eta(t)$ converges to a desired position and attitude η_d (Pose stabilization)

Model:

$$\begin{split} M \, \dot{\nu} + C(\nu) \, \nu + D(\nu) \, \nu + g(\eta) &= \tau \\ \dot{\eta} &= J(\eta) \, \nu \end{split}$$

Error Dynamics:

$$e(t) = \eta(t) - \eta_d(t) \longrightarrow \dot{\eta} = \dot{\eta} = J(\eta)\nu$$

Control Lyapunov Function (CLF):

$$V(\nu, e) = \frac{1}{2} \left(\nu^T M \nu + e^T K_P e \right)$$

Computing the time derivative with respect to the trajectory of the system...

$$\begin{split} \dot{V} &= \nu^T M \dot{\nu} + \dot{e}^T K_P e \\ &= \nu^T \left[M \dot{\nu} + J^T(\eta) K_P e \right] \\ &= \nu^T \left[\tau - D(\nu) \nu - g(\eta) + J^T(\eta) K_P e \right] - \nu^T C(\nu) \nu \end{split}$$

Assign the feedback law...

$$\tau = -J^T K_P e(t) - K_D \nu + g(\eta)$$

and we obtain

$$\dot{V} = -\nu^T \left[D(\nu) + K_D \right] \nu \le 0$$

Thus the origin $(e, \nu) = 0$ is stable.

Can we prove Asymptotic Stability ?

Use LaSalle's invariance principle...

$$E = \{(\nu, e) \in \mathbb{R}^{12} : \nu = 0\} \quad \longrightarrow \quad 0 = J^T(\eta) K_P e^{-t}$$

The largest invariant set M in E is the origin, thus we have asymptotic stability!

Therefore,

$$\lim_{t \to \infty} \eta(t) = \eta_d$$

Lyapunov redesign

Consider the system

$$\dot{x} = f(t,x) + G(t,x)[u + \delta(t,x,u)]$$

where $x\in\mathbb{R}^n,\,u\in\mathbb{R}^p$ and $\delta(\cdot)$ is an unknown disturbance that may depend on time, state, and input.

Suppose that for the nominal system

$$\dot{x} = f(t, x) + G(t, x)u$$

we have suceeded to design a feedback control law

$$u = \psi(t, x)$$

such that the origin x = 0 of

$$\dot{x} = f(t, x) + G(t, x)\psi(t, x)$$

is GUAS.

Lyapunov redesign

Furthermore, suppose that we have a C^1 function V(t,x) that satisfies

$$\begin{aligned} \alpha_1(\|x\|) &\leq V(t,x) \leq \alpha_2(\|x\|) \\ \frac{\partial V}{\partial t} + \frac{\partial V}{\partial x} [f(t,x) + G(t,x)\psi(t,x)] \leq -\alpha_3(\|x\|) \end{aligned}$$

where $\alpha_1, \alpha_2 \in \mathcal{K}_{\infty}$ and $\alpha_3 \in \mathcal{K}$. Assume that with $u = \psi(t, x) + v$ the disturbance term δ satisfies

$$\|\delta(t, x, \psi(t, x) + v)\| \le \rho(t, x) + k_0 \|v\|, \quad 0 \le k_0 < 1$$

where ρ is a nonnegative continuous function that estimates the size of the disturbance. Note that this is the only information about δ that we need to know.

Goal: Design an additional feedback control for v such that the overall control $\overline{u=\psi(t,x)+v}$ stabilizes the actual system. The design of v is called Lyapunov redesign.

Closed-loop system:

$$\begin{split} \dot{x} &= f(t,x) + G(t,x) \left(\psi(t,x) + v + \delta(t,x,\psi(t,x) + v) \right) \\ &= f(t,x) + G(t,x) \psi(t,x) + G(t,x) \left(v + \delta(t,x,\psi(t,x) + v) \right) \end{split}$$

Thus

$$\begin{split} \dot{V} &= \frac{\partial V}{\partial t} + \frac{\partial V}{\partial x} [f(\cdot) + G(\cdot)\psi(\cdot)] + \frac{\partial V}{\partial x} G(\cdot)[v + \delta(\cdot)] \\ &\leq -\alpha_3(||x||) + w^T v + w^T \delta \end{split}$$

where $w = \frac{\partial V}{\partial x} G(\cdot)$. Goal: Make $w^T v + w^T \delta \leq 0$.

To this effect note that

$$\begin{split} w^T v + w^T \delta &\leq w^T v + \|w\| \|\delta\| \\ &\leq w^T v + \|w\| [\rho(t,x) + k_0 \|v\| \end{split}$$

$$w^T v + w^T \delta \le w^T v + ||w|| [\rho(t, x) + k_0 ||v||]$$

Let

$$v = \begin{cases} -\eta(t,x)\frac{w}{\|w\|}, & \eta(t,x)\|w\| \ge \varepsilon \\ -\eta^2(t,x)\frac{w}{\varepsilon}, & \eta(t,x)\|w\| < \varepsilon \end{cases}$$

with $\eta(t, x) \ge 0$.

Then for $\eta(t,x)\|w\| \ge \varepsilon$ we have

$$w^{T}v + w^{T}\delta \leq -\eta(\cdot)\frac{\|w\|^{2}}{\|w\|} + \rho(\cdot)\|w\| + k_{0}\eta(\cdot)\frac{\|w\|}{\|w\|}\|w\|$$
$$= -\eta(\cdot)[1-k_{0}]\|w\| + \rho(\cdot)\|w\|$$

Choosing $\eta(t,x) \geq \frac{\rho(t,x)}{1-k_0}$ we obtain

 $w^Tv + w^T\delta \leq -\rho(\cdot)\|w\| + \rho(\cdot)\|w\| = 0$

Thus

$$\dot{V} \le -\alpha_3(\|x\|)$$

For $\eta(t,x)\|w\| < \epsilon$, we have

$$\begin{split} w^{T}v + v^{T}\delta &\leq -\eta^{2}\frac{\|w\|^{2}}{\epsilon} + \|w\|\rho + \eta^{2}k_{0}\frac{\|w\|^{2}}{\epsilon} \\ &= -(1-k_{0})\frac{\eta^{2}}{\epsilon}\|w\|^{2} + \|w\|\rho \\ &\leq (1-k_{0})[-\frac{\eta^{2}}{\epsilon}\|w\|^{2} + \eta\|w\|] \\ &= \left(\frac{1-k_{0}}{\epsilon}\right)\left(\eta\|w\| - \frac{\epsilon}{2}\right)^{2} + \frac{(1-k_{0})}{\epsilon}\frac{\epsilon^{2}}{4} \end{split}$$

Thus,

$$\dot{V} \le -\alpha_3(||x||) + \frac{\epsilon}{4}(1-k_0).$$

We can now conclude that the solution is GUUB (Globally Uniformly Ultimated Bounded). That is,

$$\exists_{T \ge t_0 \ge 0} : \|x(t)\| \le \beta(\|x(t)\|, t - t_0), \ \forall t_0 \le t < T$$

and for $t\geq T,$ $\|x(t)\|\leq\alpha(\epsilon).$ Note that if $\epsilon\to 0$ then $x\to 0.$ In particular, we can have GUAS if there is a ball $B_a=\{\|x\|\leq a\}$ such that

$$egin{aligned} &lpha_3(\|x\|) \ge \phi^2(x), \ \phi(x) > 0 \ &\eta(t,x) \ge \eta_0 > 0 \ &
ho(t,x) \le
ho_1 \phi(x) \end{aligned}$$

Then

$$\begin{split} \dot{V} &\leq -\alpha_3(\|x\|) - \frac{\eta^2}{\epsilon} (1 - k_0) \|w\|^2 + \rho \|w\| \\ &\leq -\frac{1}{2} \alpha_3(\|x\|) - \frac{1}{2} \phi^2(x) - \frac{\eta_0^2}{\epsilon} (1 - k_0) \|w\|^2 + \rho_1 \phi(x) \|w\| \\ &\leq -\frac{1}{2} \alpha_3(\|x\|) - \frac{1}{2} \begin{bmatrix} \phi(x) \\ \|w\| \end{bmatrix}^T \begin{bmatrix} 1 & -\rho_1 \\ -\rho_1 & 2\eta_0^2 \frac{(1 - k_0)}{\epsilon} \end{bmatrix} \begin{bmatrix} \phi(x) \\ \|w\| \end{bmatrix}$$

In the case that

$$\epsilon < \frac{2\eta_0^2 (1 - k_0)}{\rho_1^2}$$

we have

$$\dot{V} \le -\frac{1}{2}\alpha_3(||x||) < 0$$

and therefore x = 0 is GUAS.

Nonlinear Damping

Consider now the same system

$$\dot{x} = f(t, x) + G(t, x)[u + \delta(t, x, u)]$$

but with $\delta(t,x,u) = \Gamma(t,x)\delta_0(t,x,u)$, that is,

$$\dot{x} = f(t, x) + G(t, x)[u + \Gamma(t, x)\delta_0(t, x, u)]$$

where $\Gamma(\cdot)$ is known and δ_0 is bounded by $\|\delta_0(\cdot)\| \le k_0$.

Let

$$u = \psi(t, x) + v$$

then

$$\dot{V} = \frac{\partial V}{\partial t} + \frac{\partial V}{\partial x} [f(\cdot) + G(\cdot)\psi(\cdot)] + \frac{\partial V}{\partial x} G(\cdot)[v + \Gamma(\cdot)\delta_0(\cdot)]$$

$$\leq -\alpha_3(||x||) + w^T(v + \Gamma\delta_0)$$

where $\alpha_3 \in \mathcal{K}_{\infty}$.

$$\dot{V} \le -\alpha_3(\|x\|) + w^T (v + \Gamma \delta_0)$$

Set

 $v = -kw \|\Gamma(t,x)\|^2, \quad k > 0 \quad \longleftarrow \quad \text{nonlinear damping}$

which yields

$$\begin{split} \dot{V} &\leq -\alpha_3(\|x\|) - k\|w\|^2 \|\Gamma(\cdot)\|^2 + \|w\| \|\Gamma(\cdot)\|k_0\\ &= -\alpha_3(\|x\|) + \frac{k_0^2}{4k}\\ &= (1-\theta)\alpha_3(\|x\|) - \theta\alpha_3(\|x\|) + \frac{k_0^2}{4k}, \quad 0 < \theta < 1\\ &\leq -(1-\theta)\alpha_3(\|x\|), \quad \forall \|x\| \geq \alpha_3^{-1}(\frac{k_0^2}{4k\theta}) \end{split}$$

Thus it follows that the solutions are global uniformly ultimated bounded (GUUB).

Set

then

then

Example

 $\dot{x} = x^2 + u + x\delta_0(t)$ $V = \frac{1}{2}x^{2}$ $\dot{V} = x[x^2 + u + x\delta_0(t)].$ $u = -x^2 - x + v$ $\dot{V} = -x^2 + x[v + x\delta_0].$ Set $(v = -kw \|\Gamma\|^2)$ $v = -xx^2 = -x^3$ ~

$$\dot{V} = -x^2 - x^4 - x^2 \delta_0 = -x^2 - (x^2 - \frac{1}{2}\delta_0)^2 + \frac{\delta_0^2}{4} \le -x^2 + \frac{\delta_0^2}{4}$$

Note that the closed-loop system

$$\dot{x} = -x - x^3 + x\delta_0(t)$$

has a bounded solution no matter how large δ_0 is, and this thanks to the nonlinear damping $-x^3$.

~

Backstepping

Consider the system

$$\begin{split} \dot{\eta} &= f(\eta) + g(\eta) \xi \\ \dot{\xi} &= u \end{split}$$

where $\eta \in \mathbb{R}^m$ and $\xi \in \mathbb{R}$ is viewed as a virtual input.

Suppose the first subsystem can be stabilized by a smooth state feedback law

 $\xi = \phi(\eta)$

with $\phi(0) = 0$ that is, the origin of

 $\dot{\eta} = f(\eta) + g(\eta) \phi(\eta)$

is GAS. Suppose further that we know a C^1 Lyapunov function $V(\eta)$ that satisfies

$$\frac{\partial V}{\partial \eta} [f(\eta) + g(\eta)\phi(\eta)] \le -W(\eta)$$

Adding and subtracting $g(\eta)\phi(\eta)$,

$$\dot{\eta} = [f(\eta) + g(\eta)\phi(\eta)] + g(\eta)[\xi - \phi(\eta)]$$

Consider the change of variables

$$z = \xi - \phi(\eta)$$

then

$$\begin{split} \dot{\eta} &= [f(\eta) + g(\eta)\phi(\eta)] + g(\eta)z\\ \dot{z} &= u - \dot{\phi}\\ \dot{\phi} &= \frac{\partial \phi}{\partial \eta} [f(\eta) + g(\eta)\xi] \end{split}$$

Defining

 $v = u - \dot{\phi}$

yields

$$\dot{\eta} = [f(\eta) + g(\eta)\phi(\eta)] + g(\eta)z$$

$$\dot{z} = v$$

similar to the first system but now the origin of the first subsystem is GAS.

Consider the composite Lyapunov function

$$V_c = V(\eta) + \frac{1}{2}z^2$$

$$\begin{split} \dot{V}_c &= \frac{\partial V}{\partial \eta} [f(\eta) + g(\eta)\phi(\eta)] + \frac{\partial V}{\partial \eta} g(\eta)z + zv\\ &\leq -W(\eta) + [\frac{\partial V}{\partial \eta} g(\eta) + v]z \end{split}$$

Choosing

$$v = -\frac{\partial V}{\partial \eta}g(\eta) - kz, \quad k > 0$$

yields

$$\dot{V}_c \le -W(\eta) - kz^2 < 0$$

Since $\xi=z+\phi(\eta)$ and $\phi(0)=0$ then $(\eta=0,z=0)$ is GAS with

$$\begin{split} u &= v + \dot{\phi} \\ &= -\frac{\partial V}{\partial \eta} g(\eta) - k[\xi - \phi(\eta)] + \frac{\partial \phi}{\partial \eta} [f(\eta) + g(\eta)\xi] \end{split}$$

Example

$$\dot{x}_1 = x_1^2 - x_1^3 + x_2$$

 $\dot{x}_2 = u$

$$V_1 = \frac{1}{2}x_1^2$$

$$\dot{V}_1 = x_1(x_1^2 - x_1^3 + x_2)$$

= $-x_1^4 + x_1(x_1^2 + x_2)$

 ${\it Remark:}$ do not cancel $-x_1^4$ since it provides nonlinear damping. Choose

$$x_2 = \phi(x_1) = -x_1^2 - x_1$$

then

$$\dot{V}_1 = -x_1^4 - x_1^2 \le -x_1^2$$

which implies that $x_1 = 0$ is GES.

 $z_2 = x_2 - \phi(x_1) = x_2 + x_1^2 + x_1$

Consider

$$V_2 = V_1 + \frac{1}{2}z_2^2 = \frac{1}{2}x_1^2 + \frac{1}{2}z_2^2$$

$$\dot{V}_2 = x_1(-x_1^3 - x_1 + z_2) + z_2\left(u + (2x_1 + 1)(-x_1^3 - x_1 + z_2)\right)$$

= $-x_1^4 - x_1^2 + z_2\left(u + (2x_1 + 1)(-x_1^3 - x_1 + z_2) + x_1\right)$

Taking

$$u = -(2x_1 + 1)(-x_1^3 - x_1 + z_2) - x_1 - z_2$$

yields

$$\dot{V}_2 = -x_1^4 - x_1^2 - z_2^2$$

which implies that x = 0 is GAS.

Let

set

Example: Third-order system

$$\begin{aligned} \dot{x}_1 &= x_1^2 - x_1^3 + x_2 \\ \dot{x}_2 &= x_3 \\ \dot{x}_3 &= u \\ V_2 &= \frac{1}{2}x_1^2 + \frac{1}{2}z_1^2 \\ \dot{V}_2 &= -x_1^2 - x_1^4 - z_2^2 + z_2 z_3 \\ V_3 &= V_2 + \frac{1}{2}z_3^2 \\ \dot{V}_3 &= -x_1^2 - x_1^4 - z_2^2 + z_3[u - \dot{\phi} + z_2] \\ u &= \dot{\phi} - z_2 - z_3 \end{aligned}$$

$$\dot{V}_3 = -x_1^2 - x_1^4 - z_2^2 - z_3^2 < 0$$

then x = 0 is GAS.

Recursive application of the backstepping

Strict-feedback form

$$\begin{split} \dot{x}_1 &= x_2 + f_1(x_1) \\ \dot{x}_2 &= x_3 + f_2(x_1, x_2) \\ \vdots \\ \dot{x}_i &= x_{i+1} + f_i(x_1, x_2, ..., x_i) \\ \vdots \\ \dot{x}_n &= f_n(x_1, x_2, ..., x_n) + u \end{split}$$

Idea: Consider the state $x_2 \ {\rm as} \ {\rm a}$ virtual control input for $x_1 \ {\rm and}$ if it was a real input set

$$x_2 = -x_1 - f_1(x_1)$$

$$V_1 = \frac{1}{2}x_1^2 \quad \longrightarrow \quad \dot{V}_1 = -x_1^2$$

Define

$$z_1 = x_1$$

 $z_2 = x_2 - \alpha(x_1), \quad \alpha(x_1) = -x_1 - f_1(x_1)$

Then

$$\begin{split} \dot{z}_1 &= -z_1 + z_2 \\ \dot{z}_2 &= x_3 + f_2(x_1, x_2) - \frac{\partial \alpha_1}{\partial x_1}(x_2 + f_1(x_1)) = x_3 + \bar{f}_2(z_1, z_2) \\ V_1 &= \frac{1}{2}z_1^2 \\ \dot{V}_1 &= -z_1^2 + z_1 z_2 \end{split}$$

Next step:

$$\begin{aligned} z_3 &= x_3 - \alpha_2(z_1, z_2) \\ V_2 &= V_1 + \frac{1}{2} z_2^2 \\ \dot{V}_2 &= -z_1^2 + z_2 \big(z_1 + z_3 + \alpha_2(z_1, z_2) + \bar{f}_2(z_1, z_2) \big) \end{aligned}$$

$$\dot{V}_2 = -z_1^2 + z_2 (z_1 + z_3 + \alpha_2(z_1, z_2) + \bar{f}_2(z_1, z_2))$$

Choose $lpha_2(z_1,z_2)=-z_1-z_2-ar{f}_2(z_1,z_2)$ yields

$$\dot{z}_1 = -z_1 + z_2 \dot{z}_2 = -z_1 - z_2 + z_3 \dot{V}_2 = -z_1^2 - z_2^2 + z_2 z_3$$

*i*th step:

$$z_{i+1} = x_{i+1} - \alpha_i(z_1, ..., z_i)$$
$$V_i = \frac{1}{2}(z_1^2 + z_2^2 + ... + z_i^2)$$

$$\begin{aligned} \dot{z}_i &= z_i + \alpha_i(z_1, ..., z_i) + \bar{f}_i(z_1, ..., z_i) \\ V_i &= -z_1^2 - z_2^2 - ... - z_i^2 + z_{i-1} z_i + z_i \left(z_{i+1} + \alpha_i(z_1, ..., z_i) + \bar{f}_i(z_1, ..., z_i) \right) \end{aligned}$$

Using
$$\alpha(\cdot)=-z_{i-1}-z_i-\bar{f}_i(\cdot)$$
 it follows that
$$\dot{z}_i=-z_{i-1}-z_i+z_{i+1}$$
$$\dot{V}_i=-z_1^2-\ldots-z_i^2+z_iz_{i+1}$$

$$\dot{z}_n = \bar{f}_n(z_1, \dots, z_n) + u$$

Choose

$$u = \alpha_n(\cdot) = -z_{n-1} - z_n - \bar{f}_n(\cdot)$$

$$V_n = \frac{1}{2}(z_1^2 + \dots + z_n^2)$$

$$\dot{z}_n = -z_{n-1} - z_n$$

 $\dot{V}_n = -z_1^2 - z_2^2 - \dots - z_n^2$

Example

$$\dot{\eta} = \eta^2 - \eta \xi$$
$$\dot{\xi} = u$$

$$V_{1} = \frac{1}{2}\eta^{2} \longrightarrow \dot{V}_{1} = \eta(\eta^{2} - \eta\xi)$$

if $\phi = \eta - \eta^{2}$, $z_{2} = \xi - \phi$
 $\dot{V}_{1} = -\eta^{4} - \eta^{2}z_{2}$
 $V_{2} = \frac{1}{2}\eta^{2} + \frac{1}{2}z_{2}^{2}$
 $\dot{V}_{2} = -\eta^{4} - \eta^{2}z_{2} + z_{2}[u - (\eta^{2} - \eta\xi) - 2\eta(\eta^{2} - \eta\xi)]$
Choosing $u = (1 + 2\eta)(\eta^{2} - \eta\xi) - kz_{2} + \eta^{2}$ we obtain
 $\dot{V}_{2} = -\eta^{4} - kz_{2} < 0$

Block Backstepping

Consider the system

$$\begin{split} \dot{\eta} &= f(\eta) + G(\eta)\xi \\ \dot{\xi} &= f_a(\eta,\xi) + G_a(\eta,\xi)u \end{split}$$

with $\eta \in \mathbb{R}^n$, $\xi \in \mathbb{R}^m$, $u \in \mathbb{R}^m$ and $G_a(\cdot)$ nonsingular.

Suppose that there exist $\phi(\eta)$, $\phi(0) = 0$ and $V(\eta)$ that satisfies

$$\frac{\partial V}{\partial \eta} [f(\eta) - G(\eta)\phi(\xi)] \le -W(\eta)$$

Using

$$V_c = V(\eta) + \frac{1}{2} [\xi - \phi(\eta)]^T [\xi - \phi(\eta)]$$
$$\dot{V}_c = \frac{\partial V}{\partial \eta} (f + G\phi) + \frac{\partial V}{\partial \eta} G(\xi - \phi) + (\xi - \phi)^T [f_a + G_a u - \frac{\partial \phi}{\partial \eta} (f + G\xi)]$$

Taking

$$u = G_a^{-1} \left[\frac{\partial \phi}{\partial \eta} (f + G\xi) - \left(\frac{\partial V}{\partial \eta} G \right)^T - f_a - k(\xi - \phi) \right], \ k > 0$$

results in

$$\dot{V}_c \le -W(\eta) - k[\xi - \phi(\eta)]^T [\xi - \phi(\eta)] < 0$$

and so $[\eta, \xi]^T = 0$ is GAS.